Peridynamic modeling of rail squats
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ABSTRACT: Rail squats and studs are typically classified as the propagation of any cracks that have grown longitudinally through the subsurface. Some of the cracks could propagate to the bottom of rails transversely, which have branched from the initial longitudinal cracks with a depression of rail surface. The rail defects are commonly referred to as ‘squats’ when they were initiated from damage layer caused by rolling contact fatigue, and as ‘studs’ when they were associated with white etching layer caused by the transform from pearlitic steel due to friction heat generated by wheel sliding or excessive traction. Such above-mentioned rail defects have been often observed in railway tracks catered for either light passenger or heavy freight traffics and for low, medium or high speed trains all over the world for over 60 years except some places such as sharp curves where large wear takes place under severe friction between wheel flange and rail gauge face. It becomes a much-more significant issue when the crack grows and sometimes flakes off the rail (by itself or by insufficient rail grinding), resulting in a rail surface irregularity. Such rail surface defect induces wheel/rail impact and large amplitude vibration of track structure and poor ride quality. In Australia, Europe and Japan, rail squats/studs have occasionally turned into broken rails. The root cause and preventive solution to this defect are still under investigation from the fracture mechanics and material sciences point of view. Some patterns of squat/stud development related to both of curve and tangent track geometries have been observed, and squat growth has also been monitored for individual squats using ultrasonic mapping techniques. This paper highlights peridynamic modeling of squat/stud distribution and its growth. Squat/stud growth has been measured in the field using the ultrasonic measurement device on a grid applied to the rail surface. The depths of crack paths at each grid node form a three dimensional contour of rail squat crack. The crack propagation of squats/studs is modelled using peridynamics. The modeling and field data is compared to evaluate the effectiveness of peridynamics in modelling rail squats.

INTRODUCTION
Rail squats, defined as cracks initiated from rolling contact fatigue (RCT) and from white etching layer (WEL); and growing longitudinally under the rail surface (in the direction to train), are a main problem for rail operators all around the globe. They are noticed by passengers when they create excess noise and vibration leading to uncomfortable rides (Remennikov and Kaewunruen 2008), but more importantly they can result in broken rail from the impact forces from wheel-rail interaction (Kaewunruen and Remennikov 2010, 2009). Additionally, squats have grown and
turned into broken rails, which could result in a catastrophe (Ishida 2013). In practice, the rail surface defects have been a critical safety concern and key maintenance priority of railway infrastructure owners and managers who operate either low, moderate or high speed trains including passenger suburban, metro, urban, mixed-traffic and freight rail systems. The rail surface defects can cause high risks and significant consequences such as train derailments from rail breaks, component failures, and so on. It has been estimated that the cost of rail renewal program (rail replacement) due to rail squats and studs has become a significant portion of the whole track maintenance cost, reportedly in Australia, Asia, and European countries e.g. Austria, Japan, Germany and France (Kaewunruen et al., 2015; Kaewunruen and Ishida, 2014; 2015; 2016). The rail squat/stud problem has largely been noticed when the ride quality of the passenger trains exceeds acceptable limits (Kaewunruen and Remennikov, 2016). Excessive noise and vibration have later increased complaints against rail operators. Most importantly, the impact forces due to the wheel/rail interaction have undermined the structural integrity and stability of track components (Remennikov and Kaewunruen, 2008). The study into innovative solutions for this problem is timely and significant. This paper is the world first to adopt peridynamic theory to predict dynamic crack growth from RCT, and to the best of authors’ knowledge no other such work exists currently (Kaewunruen, 2015; 2018).

Classical mechanics theory uses spatial derivatives that do not exist in when the displacement field is discontinuous, so the techniques of fracture mechanics are used to study cracking phenomena. However, a major drawback is that crack path must be known a priori. Therefore, peridynamics (PD) (Silling 2000; Silling et al. 2007) was created to simulate objects with discontinuities. It uses integral not partial-differential equations, and deformation instead of strain to compute the internal forces. Since both are defined in the presence of cracks, PD are well suited for such analyses. Moreover, in PD crack path doesn’t have to be known –cracks initiate automatically according to a prescribed damage law. These reasons make PD an excellent tool for studying different kinds of fracture and have been used to study damage in fiber-reinforced laminated composites (Colavito 2013; Hu, De Carvalho, and Madenci 2015; Hu, Madenci, and Phan 2017; Kilic, Agwai, and Madenci 2009), glass (Bobaru, Ha, and Hu 2012; Bobaru and Zhang 2016), wood (Perré et al. 2015), concrete (Gerstle, Sau, and Sakhavand 2009; Shen, Zhang, and Huang 2013; Yaghoobi and Chorzepea 2015) and steel (De Meo et al. 2016).

STATE-BASED PERIDYNAMICS
This section presents a brief overview of the state-based peridynamics theory, for an extended overview authors recommend (Bobaru et al. 2016; Madenci and Oterkus 2014; Silling and Lehoucq 2010). A peridynamic body consists of a number of nodes in the reference position $\mathbf{x}_i$ each describing some volume $V_{\mathbf{x}_i}$. A node $\mathbf{x}_i$ interacts with other nodes $\mathbf{x}_j$ within a range called the horizon $\delta_i$ through bonds. Nodes within this range is called the family of $\mathbf{x}_i, H_{\mathbf{x}_i}$. When a body undergoes some deformation, node $\mathbf{x}_i$ experiences displacement $\mathbf{u}_i$ and moves to a deformed position $\mathbf{y}_i$. This deformation creates a force density vector $t_{ij}$ that depends on the collective deformation of $H_{\mathbf{x}_i}$ and $t_{ji}$ that depends on the collective deformation of $H_{\mathbf{x}_j}$. The bond deformation vectors
and the force density vectors are stored in arrays called the deformation states, $Y_{xi}$ and the force states $T_{xi}$:

$$
Y_{xi} = \begin{bmatrix} y_1 - y_i \\ \vdots \\ y_n - y_i \end{bmatrix},
T_{xi} = \begin{bmatrix} t_{1i} \\ \vdots \\ t_{ni} \end{bmatrix}.
$$

(1)

It is common to write $T(x_i)\langle x_j - x_i \rangle$ when referring to a force density vector $t_{ij}$ in a specific bond $\xi_{ij} = x_j - x_i$. The force density vectors depend on the deformation, so we can write:

$$
T(x_i) = T(Y(x_i))
$$

(2)

where the function $T(x_i)$ is the material model. Then the peridynamic equation of motion in the integral form is

$$
\rho(x_i) \ddot{u}(x_i, t) = \int_{H_{xi}} (T(x_i)\langle x_j - x_i \rangle - T(x_j)\langle x_j - x_i \rangle) dV_{ij} + b(x_i)
$$

(3)

where $\rho(x_i)$ – density, $\ddot{u}(x_i)$ – acceleration and $b(x_i)$ – external force density.

Boundary conditions are not required in PD solution, because the PD equation of motion does not contain any spatial derivatives, however, they are necessary to solve many real-life problems. Since nodes describe some volume, boundary conditions must also be applied to some volume.

Damage is introduced by breaking a bond. The simplest damage criterion could be the critical stretch, in which a bond breaks when it’s stretched past some critical value $s_c$:

$$
\omega(x_i) = \begin{cases} 1, & \text{if } s_{ij} < s_c \\ 0, & \text{if } s_{ij} \geq s_c \end{cases}, \quad s_{ij} = \frac{\|y_j - y_i\| - \|x_j - x_i\|}{\|x_j - x_i\|} = \frac{\|Y(\xi_{ij})\| - \|\xi_{ij}\|}{\|\xi_{ij}\|}.
$$

(4)

The damage at a node is defined in (Silling and Askari 2005) as a ratio between the broken and the initial number of bonds:

$$
\phi(x_i) = 1 - \frac{\int_{H_{xi}} \omega(x_i) dV_{ij}}{\int_{H_{xi}} dV_{ij}}.
$$

(5)

The PD fatigue damage model used in this study was introduced in (Silling and Askari 2014) and used in (Jung and Seok 2017, 2016; Zhang et al. 2016). Other researchers have also developed fatigue damage models (Baber and Guven 2017; Oterkus, Guven, and Madenci 2010), however, these models use bond-based PD and simulate only the crack growth phase. The overview of the model is given in (Silling and Askari 2014), but it’s repeated here for completeness.

A body undergoes some cyclic deformation between two extremes + and -, bond strains at each extreme and the cyclic bond strain is:

$$
\sigma_{ij} = \frac{|Y^+(\xi_{ij}) - \|\xi_{ij}\|}{\|\xi_{ij}\|}, \quad \sigma_{ij} = \frac{|Y^-(\xi_{ij}) - \|\xi_{ij}\|}{\|\xi_{ij}\|}, \quad \epsilon_{ij} = |\sigma_{ij} - \sigma_{ij}|.
$$

(6)

For each bond a variable called the “remaining life” $\lambda_{ij}(x_i, \xi_{ij}, N)$ is defined. It degrades at each loading cycle $N$, and a bond breaks when the remaining life is reduced to zero:

$$
\lambda_{ij}(N) \leq 0.
$$

(7)

At the beginning when $N = 0$:

$$
\lambda_{ij}(0) = 1.
$$

(8)

then at each cycle in crack nucleation phase (phase I) the change of $\lambda$ is given by
\[
\frac{d\lambda_{ij}}{dN}(N) = \begin{cases} 
-A_f(e_{ij} - \varepsilon^m), & \text{if } e_{ij} > \varepsilon^m \\
0, & \text{if } e_{ij} \leq \varepsilon^m
\end{cases}
\]  
(9)

where \(\varepsilon^m\) - fatigue limit under which no fatigue damage occurs, \(A_f, m_1\) – parameters for phase I. In the phase II the remaining life changes according to:

\[
\frac{d\lambda_{ij}}{dN}(N) = -A_I \varepsilon_{ij}^{m_{II}}. 
\]  
(10)

The transition from phase I to phase II is handled by applying the phase I model with parameters \(A_f, m_1\) to a node \(x_t\) till there is a node \(x_j\) in \(H_{x_t}\) with damage \(\phi(x_j) \geq 0.5\), then reset the remaining life of the bonds connected to \(x_j\) to 1 and switch to phase II model.

**COMPUTATIONAL MODEL**

The fatigue damage model was implemented in the open-source Peridynamics program Peridigm (Lihlewood et al. 2013; Parks et al. 2012). For quasi-static analysis the acceleration is zero, so peridynamic equation of motion is approximated as:

\[
\sum_{H_{x_t}} (T[x_t,t](x_j - x_i) - T[x_j,t](x_i - x_j)) \Delta V_{x_j} + b(x_i,t) = 0.
\]  
(12)

and then solved using Newton’s method. The remaining life of each bond is computed after each simulation step, by:

\[
\lambda^0_{ij} = 1, \quad \lambda^N_{ij} = \lambda^{N-1}_{ij} - \Delta N A(\varepsilon^N_{ij})^m. 
\]  
(13)

(Silling and Askari 2014) introduces two techniques – implicit strain simulation and time mapping – to speed up simulations. Both were used and are repeated here for completeness. In case of a high-cycle fatigue the bond strains from the train wheel load are below the elastic limit, so an elastic material model is used to simulate the rail. In such a case strain in a bond would change linearly between + and – loading conditions, so it is possible to simulate only the + loading condition and compute

\[
s^- = R s^+, \quad R = \frac{P^-}{P^+}. 
\]  
(14)

where \(R\) – loading ratio, \(P\) – applied load at each extreme. The cyclic strain is then given by:

\[
\varepsilon = |s^+ - s^-| = |(1 - R)s^+|. 
\]  
(15)

Simulation time is mapped against the current cycle using a linear mapping:

\[
N = \frac{t}{\tau}, 
\]  
(16)

where \(\tau\) is a constant. Then remaining life change in current simulation time is found through:

\[
\frac{\Delta \lambda}{\Delta t} = \frac{\Delta \lambda}{\Delta N} \frac{\Delta N}{\Delta t}. 
\]  
(17)

**MODEL OF A RAIL**

The model was discretized using meshless method introduced in (Silling and Askari
Each node has a position in 3D space and describes some amount of volume around it. The edge length of a node was 0.001m and the horizon $\delta = 0.00201$m. The mesh is rather coarse and the horizon rather short, these values were selected to reduce computational expense, because here the preliminary results of a larger work are presented. In this study a model of UIC60 profile rail head was used. Due to irregular form of the rail head, nodes were not perfectly cubic; however, differences were insignificant near the middle of the rail head and increased only slightly near the gauge corners. The shape of the rail head was first created with solid hexahedral elements in finite element (FE) program Ansys, then element centroids and volumes were exported to a text file to be used as a mesh for Peridigm. Movement in vertical (y) direction was fixed for a layer of nodes with thickness $\delta$ at the bottom of the rail head, additionally damage was forbidden for nodes less than $3\delta$ from bottom, to avoid unphysical behavior near the boundary conditions.

Since applied loads didn’t cause the material to exceed its yield strength, an elastic material model was used, it’s properties: density – 7850 kg/m$^3$, Poisson’s ratio – 0.3, Young’s modulus – 210 GPa. The instructions on how to obtain damage model parameter values are included in (Silling and Askari 2014), they were applied to test data presented in (Scutti et al., 1984). While the data were quite old, they were used, because all four fatigue model parameters could be estimated from just one data source, since both E-N curves and Paris law plots are presented. Fatigue damage model parameters were: $A_1 = 426.00$, $m_1 = 2.77$, $A_2 = 3249.00$, $m_2 = 4.00$. The model switched from phase I to phase II, when damage at a node reached 40%. Also, a linear time mapping was used, with $\tau$ equal to 0.001 till 21’000 cycles, 0.01 till 26’200 cycles, and 0.1 till 26670th cycle.

**TRAIN WHEEL LOADING**

Two different train wheel loadings were applied – vertical pressure and surface shear traction. They were obtained from (Wei et al. 2016) Fig. 5f and 6f, which show elastic pressure and elastic surface shear traction respectively. Loads were applied to a single layer of nodes at the middle of rail head’s top surface, with a loading ratio $R = 0$. The wheel-rail contact area (Fig. 4f in (Wei et al. 2016)) was approximated with an ellipse whose half-axis were $a=0.0066$m, $c = 0.006386$m. The elastic pressure was approximated by an ellipsoid with half-axis $a=0.0066$m, $c = 0.006386$m, $b = 1.116$GPa, then the value at each node was calculated from ellipsoid formula and converted to force density. Shear traction forces change based on the location of a node in the contact area, so they were described by four tri-linear functions (Q1 to Q4). First, nodes were split into seven intervals based on their transversal location (z coordinate), second a tri-linear shear traction function was selected based on the interval, finally shear traction value was calculated and converted to force density. Table 1 shows interval limits and corresponding surface shear traction functions, Fig. 1 presents the surface shear traction functions.

**RESULTS**

Simulation ran for 26’670 cycles. Fig. 2 shows damage in the rail heads cross-section parallel to the longitudinal direction. First bonds break before 14’000$^{th}$ cycle, then
damage zone under applied loads expands till at 22’600\textsuperscript{th} cycle node reaches 40% damage (not showed in Fig. 2.), and its neighborhood gets switched to phase II. Damage develops quite rapidly after that, leading to near complete damage (80%) at a node at cycle 26’670. Most of the broken bonds are concentrated under the maximum pressure as expected, because it creates larger bond strains. This simulation shows a very fast onset of damage that doesn’t grow into longer cracks, instead damage is contained under the load area. However, it can be seen in Fig. 2 that damage extends forward under the surface forming the beginning of a squat. Longer simulations are needed to see if damage really expands in that direction. Fig. 3 shows damage in cross-section perpendicular to the longitudinal direction. The damage is nearly symmetrical at 20’000 cycles, but asymmetrical at 26’670 cycles. Most likely this is caused by the iterative solver converging to slightly different values and the mesh not being completely symmetric.

Additionally, Fig. 3 (b) shows that the top layer of nodes has broken off the following layers and penetrated them, which is an unphysical behavior. Since a large (>80%) number of bonds are broken for the top layer and loads are applied only to this layer, they no longer are distributed further down. This is the likeliest explanation for why cracks do not grow into longer squats. This problem could be remedied with finer mesh, because the same loads will be distributed over larger number of nodes, resulting in less deformation at each node and, therefore, bond. Applying loads to a thicker layer could help distribute the wheel loading in way that’s closer to real life conditions. It was suggested in (Macek and Silling 2007) that loads are applied to a layer equal to the horizon. Finally, selecting longer horizon could help distribute loading to lower layers.

**CONCLUSIONS**

Peridynamic theory can be used to simulate rolling contact fatigue damage in a railway rail. The maximum damage will occur under the maximum pressure values as expected. The simulation showed onset of a rail squat, however simulations limitations didn’t allow author’s to further explore it. To remedy this boundary conditions must be applied in a way that ensures continuous load transfer to lower layers, even after surface layer sustains large damage.

**REFERENCES**


### TABLES

Table 1. Interval limits in transverse direction and their tri-linear surface shear traction functions.

<table>
<thead>
<tr>
<th>Interval</th>
<th>Interval start transversal coordinate, m</th>
<th>Interval end transversal coordinate, m</th>
<th>Tri-linear Surface Shear Traction Function</th>
</tr>
</thead>
<tbody>
<tr>
<td>1</td>
<td>-0.006386</td>
<td>-0.005</td>
<td>Q4</td>
</tr>
<tr>
<td>2</td>
<td>-0.005</td>
<td>-0.003</td>
<td>Q3</td>
</tr>
<tr>
<td>3</td>
<td>-0.003</td>
<td>-0.001</td>
<td>Q2</td>
</tr>
<tr>
<td>4</td>
<td>-0.001</td>
<td>0.001</td>
<td>Q1</td>
</tr>
<tr>
<td>5</td>
<td>0.001</td>
<td>0.003</td>
<td>Q2</td>
</tr>
<tr>
<td>6</td>
<td>0.003</td>
<td>0.005</td>
<td>Q3</td>
</tr>
<tr>
<td>7</td>
<td>0.005</td>
<td>0.006386</td>
<td>Q4</td>
</tr>
</tbody>
</table>
FIGURES

FIG. 1. Surface shear traction (GPa) vs the x coordinate of a node (m). Dashed lines show data from (Wei et al. 2016) and solid lines show the fitted tri-linear functions (Q1 to Q4).

FIG. 2. Cross-section (plane parallel to the longitudinal direction) at the middle
of the rail head showing damage from rolling contact fatigue at (a) - 14'000 cycles, (b) – 21’000, (c) – 25’000, (d) – 26’200, (e) – 26’580, (f) – 26’670. Rolling direction is to the right and only damaged zone is shown. Blue color – less damage, red – more damage.

FIG. 3. Cross-section (plane normal to longitudinal direction) at the middle of rail head (center of the applied load) showing damage from rolling contact fatigue at (a) - 20’000 cycles, (b) – 26’670. Color scheme indicates damage at a node with blue – less damage, red – more damage.