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1 Effects of ambient pressure on ignition and flame characteristics in diesel spray combustion

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#### 13 Abstract

14 This work reports on numerical investigation of effects of ambient pressure  $(P_{am})$  on spray combustion under engine-like conditions. Three cases with different Pam of 42, 85 and 170 bar at 15 a fixed ambient temperature of 1000 K are considered. Zero-dimensional calculations are first 16 performed for autoignition of stagnant adiabatic homogenous mixtures to evaluate performance 17 of the selected diesel surrogate fuel models and to identify the  $P_{am}$  effects on the most reactive 18 mixture. An Eulerian-based transported probability density function model is then chosen for the 19 three-dimensional computational fluid dynamics study. The results show the predicted ignition 20 delay times and flame lift-off lengths are in reasonably good agreement with experiment, with the 21 relative difference below 28%. The current work reveals that low-temperature reactions occur 22 across a wide range of mixture fraction but a noticeable rise of temperature (>100 K above 23 ambient temperature) is detected first on the fuel-lean side of the stoichiometric line in all three 24 cases. The high-temperature ignition occurs first on the fuel-rich side in the 42 and 85 bar cases, 25 where the igniting mixture appears to be more fuel-rich in the latter case. As  $P_{am}$  is further 26

increased to 170 bar, the igniting mixture becomes more fuel-lean and the high-temperature 27 ignition occurs on the fuel-lean side. The ignition behavior is found to depend on both physical 28 and chemical processes. At 170 bar, the reaction rate increases and the associated transition from 29 low- to high-temperature ignition is relatively fast, as compared to the transport of warmer 30 products from the lean zone into the fuel-rich mixture. Also, within the fuel-rich region, the local 31 temperature is low due to liquid fuel vaporization and the condition is not appropriate for 32 ignition. These collectively cause the high-temperature ignition to occur on the fuel-lean side. 33 Analyses on the quasi-steady spray flame structures reveal that, apart from poorer air entrainment 34 due to reduced lift-off length, the higher rich-zone temperature and lower scalar dissipation rate 35 also lead to a higher peak soot volume fraction at higher  $P_{am}$ . 36

37

38 **Keywords:** spray flame, transported probability density function, ignition, pressure effects

39

#### 40 **1. Introduction**

Over the last few decades, automotive, light-duty, heavy-duty and marine engine industries have 41 42 been working on improving the understanding of in-cylinder phenomena, which is an important prerequisite to design clean and efficient engines [1-12]. Numerous research groups, mainly led 43 by the Sandia National Laboratory, share their optical measurements and modelling results 44 through the Engine Combustion Network (ECN) [3], with the aim to facilitate and consolidate 45 diesel spray combustion research. Effects of a wide range of diesel engine parameters on fuel 46 penetration, ignition delay time (IDT), flame lift-off length (LOL) and soot volume fraction 47  $(f_{v,soot})$  have been studied. However, the majority of these studies were performed for an ambient 48 density ( $\rho_{am}$ ) range of 14.8 to 22.8 kg/m<sup>3</sup>. In particular, the Spray A condition, where  $\rho_{am}$  is fixed 49 at 22.8 kg/m<sup>3</sup> has been broadly studied in the last couple of years, with an emphasis on 50 51 understanding the ignition and flame stabilization processes [4-7]. The initial ambient air temperature before the air is cooled by the vaporized fuel,  $T_{am}$ , is set to 900 K, corresponding to 52

an ambient pressure  $(P_{am})$  of ~67 bar. Under the typical range of  $T_{am}$  at the start of injection in 53 heavy-duty and large two-stroke marine engines, the ambient densities can be approximately 30.0 54 and 60.0 kg/m<sup>3</sup> under part and full load conditions, respectively [10,12-15]. Optical accessible 55 experiments conducted at such a high  $\rho_{am}$  (or  $P_{am}$ ) were reported by Siebers and co-workers [8-56 12]. In their experimental studies, the  $\rho_{am}$  was varied from 7.3 to 58.5 kg/m<sup>3</sup>, corresponding to a 57  $P_{am}$  of ~20 to ~170 bar at a  $T_{am}$  of 1000 K. Under these test conditions, both the liquid and vapor 58 penetration lengths decreased with increasing  $P_{am}$ . Besides that, IDT and flame LOL were found 59 to decrease with increasing  $P_{am}$ , while the spray flame size at quasi-steady state (in terms of both 60 length and width) reduced when  $P_{am}$  increased [10,11]. The effects of  $P_{am}$  on ignition and 61 premixed combustion processes up to 45.0 kg/m<sup>3</sup> (~120 bar at  $T_{am}$  of 1000 K) were also 62 investigated by Higgins et al. [12]. In terms of soot formation, measurements from optical 63 accessible experiments showed path-length-averaged  $f_{v,soot}$  and maximum  $f_{v,soot}$  increased with  $P_{am}$ 64 [3,13]. Nevertheless, it is challenging to deduce the low- and high-temperature ignition as well as 65 other combustion phenomena solely based on experimental observation at such elevated  $P_{am}$ . To 66 date, the effects of  $P_{am}$  on spray flame structures under these conditions were not explicitly 67 studied numerically, although these experimental data have been used for computational fluid 68 dynamic (CFD) model validation [16,17]. A more comprehensive understanding of  $P_{am}$  effects is 69 crucial for both modelling of engine combustion and for developing physical and chemical 70 models at these  $P_{am}$  levels. 71

Set against this background, the objectives of this CFD work are first to generate velocity, species and temperature distribution profiles to help improving the understanding of the ignition/combustion process at high  $P_{am}$  levels; and second to elucidate the effects of  $P_{am}$  on spray flame structure and emissions formation. These aims are achieved by employing an Eulerian-based transported probability density function (PDF) method [18] and a skeletal *n*heptane mechanism [19]. Zero-dimensional calculations are also performed for autoignition of stagnant adiabatic homogenous mixtures to complement the CFD study. The remainder of the paper is structured as follows. In the next section, experimental data used for model validation are first described. It is followed by the descriptions of the numerical methods. The subsequent sections detail the model validation as well as the numerical analyses of autoignition and flame structures at different  $P_{am}$  levels. Conclusions from this work are outlined in the final section.

84

#### 85 **2. Experimental data for model validation**

The grade number two diesel fuel (diesel #2) spray experimental data used for model validation 86 in this work were obtained in a constant volume pre-burn chamber [9,10]. The initial ambient gas 87 composition is determined by the composition of the combustible-gas mixture burned to generate 88 diesel engine-like temperatures and pressures. In the selected test conditions, the reaction 89  $3.245 \cdot C_2H_2 + 0.515 \cdot H_2 + 8.37 \cdot O_2 + 89.75 \cdot N_2 \rightarrow 89.75 N_2 + 6.49 \cdot CO_2 + 3.76 \cdot H_2O$  was used to 90 generate the inert high-pressure, high-temperature environment in the non-reacting spray cases. A 91 92 combustible-gas mixture of 68.1% N<sub>2</sub>, 28.4% O<sub>2</sub>, 3.0% C<sub>2</sub>H<sub>2</sub> and 0.5% H<sub>2</sub> (by volume) was then used in the reacting spray cases. The product composition of this combustible mixture simulated 93 94 air, having a composition of 21.0% O<sub>2</sub>, 69.3% N<sub>2</sub>, 6.1% CO<sub>2</sub>, and 3.6% H<sub>2</sub>O (by volume) [10-12]. The experimental initial ambient gas composition is used in the current reacting spray cases 95 in order to produce identical initial thermochemical conditions. The model performance in 96 simulating fuel penetration lengths of evaporating, non-reacting sprays (cases 1 to 3 shown in 97 Table 1) at different  $P_{am}$  is first evaluated. For liquid-phase, the comparison is made against the 98 liquid penetration length determined with the liquid length scaling law. The properties of n-99 heptadecane are used in the liquid length scaling law to produce diesel #2 liquid length [8] since 100 these resemble the properties of diesel #2. The simulated vapor penetration length is compared 101 with the measurements reported by Naber and Siebers [9]. For the validation of the reacting 102 sprays (cases 4 to 6 shown in Table 1), IDT and LOL measurements from the ECN database [3] 103 are used; more descriptions can be found in Ref. [10]. 104

105 Table 1. Operating conditions and injection specifications

Case	$O_{2,am}$	Tam	$\rho_{am}$	Dnozz	Pinj	$\dot{m}_{f}$
		[K]	$(kg/m^3)$	(µm)	(bar)	(mg/ms)
1	0	1000	13.9	257	1400	14.0
2	0	1000	28.6	257	1400	14.0
3	0	1000	58.6	257	1400	14.0
$4^*$	21%	1000	14.8	180	1400	8.8
$5^{*}$	21%	1000	30.0	180	1400	9.0
$6^*$	21%	1000	58.5	180	1400	9.2

\*Note: Non-reacting, vaporising spray simulations are also carried out for cases 4 to 6, where  $O_{2,am}$  is set to 0.

As depicted in Table 1, three cases with different  $\rho_{am}$  of 14.8, 30.0 and 58.5 kg/m<sup>3</sup> are considered for the reacting conditions. At a fixed  $T_{am}$  of 1000 K, these correspond to  $P_{am}$  of 42, 85 and 170 bar, respectively. Initial species composition, flow and turbulence conditions can be found in a previous work [20]. The operating conditions as well as the injection specifications including nozzle diameter ( $D_{nozz}$ ), injection pressure ( $P_{inj}$ ) and fuel mass flow rate ( $\dot{m}_j$ ) of the test cases are listed in Table 1.

115

#### 116 **3. Numerical methods**

#### 117 3.1 Autoignition of homogeneous mixtures

To date, effects of  $P_{am}$  on autoignition of various hydrocarbon fuels have been studied in shock 118 tube and plug flow reactor experiments as well as kinetic model simulations. Hashemi et al. [21-119 23] recently studied the effects of  $P_{am}$  on C<sub>1</sub>-C<sub>3</sub> hydrocarbon reacting pathway up to 100 bar. 120 However, for larger hydrocarbons which are commonly used as diesel #2 surrogate fuels such as 121 *n*-heptane [24,26], *n*-dodecane/myxelene [27] and diesel primary reference fuels [28], the tested 122  $P_{am}$  only considered up to ~50 bar. To the authors' knowledge, validation of diesel #2 surrogate 123 fuel kinetic models for pressure levels of 80 bar and above (targeted conditions in the current test 124 cases) is not available. Thus, five different chemical kinetic mechanisms are evaluated at 125 different Pam levels. These include the detailed Lawrence Livermore National Laboratory n-126 heptane model (the Mehl654 mechanism) [24] and two of its skeletal versions (the Liu44 and 127 Lu68 mechanisms) [19,25], a recently updated *n*-heptane model (the Zhang1268 mechanism) 128

<sup>106</sup> 107

[26], and an integrated model of *n*-dodecane and myxelene (the Pei163 mechanism) [27]. To 129 evaluate the selected chemical mechanisms, zero-dimensional (0-D) calculations are conducted 130 for autoignition of stagnant adiabatic homogenous mixtures at three different initial  $P_{am}$  of 42, 85 131 and 170 bar using the CHEMKIN-PRO software [29]. The IDT is defined as the time where the 132 mixture temperature increases to 400 K above the initial temperature ( $T_{t=0}$ ). A quadratic relation, 133  $T_{t=0} = T_{am} + 1120Z^2 - 1370Z$ , is used to correlate the temperature and mixture fraction. Here, Z is the 134 Bilger mixture fraction while  $T_{am}$  is the initial ambient temperature in the combustion chamber 135 and is set to 1000 K. This relation is obtained by post-processing the 3D-CFD results of the 136 reacting case prior to the start of low-temperature ignition, corresponding to the mixing between 137 the cold fuel and hot air. The quadratic relation is constructed for  $0 \le Z \le 0.4$  and the same 138 function is found applicable to represent the mixing at all three  $P_{am}$  (cf. Figs. 7 to 9) such that  $P_{am}$ 139 is the only variable in the comparison. Akin to that in Ref. [30], the most reactive mixture  $(Z_{mr})$  is 140 characterized as the mixture which has the shortest IDT, and the associated IDT can serve as a 141 142 reference for the comparison against the results from inhomogeneous and turbulent mixtures [30]. It may be worth mentioning that effects of pressure on  $Z_{mr}$  have not yet been explicitly studied for 143 turbulent non-premixed flames [31]. Based on the results of Fig. 1, all the models produce similar 144 qualitative trends with respect to the change of  $P_{am}$ , *i.e.*, the  $Z_{mr}$  occurs in more fuel-lean mixture 145 and  $Z_{mr}$  falls closer to the stoichiometric mixture fraction,  $Z_{st}$  when  $P_{am}$  increases (cf. Fig. 1). The 146 results also show that the IDT predicted by the Liu44 mechanism is close to those of its detailed 147 counterpart, in particular near the  $Z_{mr}$  regimes. This trend is similar to that reported by Pei et al. 148 [32] under a similar ambient, thermochemical condition, although the fuel temperature is lower at 149 373 K in their cases. 150

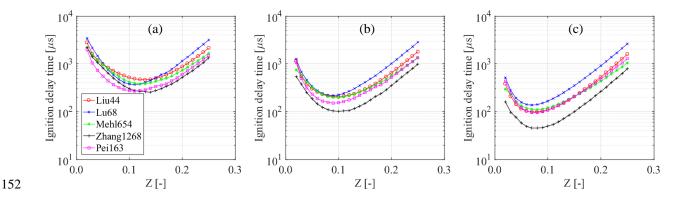
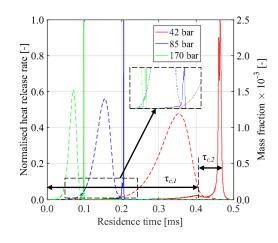


Fig. 1. Ignition delay times of homogenous mixtures for varying initial mixture fraction and temperatures at (a) 42 bar, (b) 85 bar and (c) 170 bar.



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Fig. 2. Ignition histories of  $Z_{mr}$  computed in the zero-dimensional homogenous reactor model using the Liu44 mechanism. Solid lines denote the normalized heat release rate while dotted lines represent the mass fraction of RO<sub>2</sub>.

160

In the case where the Liu44 model is used,  $Z_{mr}$  are recorded to be 0.13, 0.10 and 0.07 in 42, 85 and 170 bar case, respectively. The corresponding  $T_{t=0}$  are 841 K, 874 K and 910 K, respectively. The simulated heat-release rate (HRR) profiles are plotted in Fig. 2. Temporal evolution of the heptyl radical (RO<sub>2</sub>) of these cases are also provided to indicate the lowtemperature ignition. Here, the IDTs coincide at the peaks of the HRR. The IDT can be expressed as IDT =  $\tau_{c,1} + \tau_{c,2}$ , where  $\tau_{c,1}$  and  $\tau_{c,2}$  are the first-stage (low-temperature) and second-stage (hightemperature) IDTs respectively. All the mixtures in these cases undergo two-stage ignition, where the high-temperature ignition occurs after the high amount of RO<sub>2</sub> is consumed. It is also noted in the insets of Fig. 2 that  $\tau_{c,2}$  becomes negligible when the  $P_{am}$  increases. The Liu44 mechanism is therefore chosen for the 3-D CFD simulations, which are described next.

171

172 3.2 CFD model formulation

The 3-D CFD spray combustion simulations are carried out using OpenFOAM version 3.0.1 [33]. 173 Detailed descriptions of the model can be found in [35] and only a brief description is provided 174 here. The fuel spray, flow and combustion processes are modelled using the Eulerian-Lagrangian 175 approach within the unsteady Reynolds averaged Navier-Stokes (URANS) framework. The 176 Reitz-Diwakar model is used to simulate the fuel droplet breakup, while the Realizable  $k-\varepsilon$ 177 model is employed for turbulence modelling. For the nitric oxide (NO) formation modelling, the 178 Zeldovich thermal NO sub-mechanism from Ref. [34] are added to the Liu44 mechanism. The 179 interaction between turbulence and chemistry is modelled using the Eulerian-based transported 180 PDF method, Eulerian Stochastic Field (ESF) [18]. Similar to the Lagrangian particle transported 181 PDF method, the ESF method is a general approach to account for the turbulence-chemistry 182 interaction (TCI). It can be used to simulate the autoignition as well as different levels of 183 'premixedness' *i.e.*, premixed, partially premixed and non-premixed. In the ESF method, the 184 governing equation for the *n*-th stochastic field is 185

186

$$\overline{\rho}d\phi_{a}^{(n)} = -\overline{\rho}\widetilde{u}_{i}\frac{\partial\phi_{\alpha}^{(n)}}{\partial x_{i}}dt + \overline{\rho}S_{\alpha}^{r}(\phi^{(n)})dt + \overline{\rho}S_{\alpha}^{s}(\phi^{(n)})dt + \frac{\partial}{\rho}S_{\alpha}^{s}(\phi^{(n)})dt + \frac{\partial}{\rho}\sum_{i}\left(\Gamma_{i}\frac{\partial\phi_{\alpha}^{(n)}}{\partial x_{i}}\right)dt - \frac{1}{2}\overline{\rho}C_{\phi}(\phi_{\alpha}^{(n)} - \widetilde{\phi}_{\alpha})\omega_{i}dt + \frac{\partial}{\rho}\sqrt{2\frac{\Gamma_{i}}{\rho}}\left(\frac{\partial\phi_{\alpha}^{(n)}}{\partial x_{i}}\right)dW_{i}^{(n)}$$

$$(1)$$

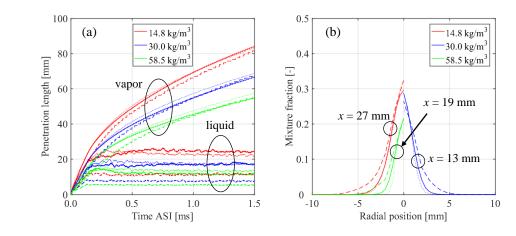
187 where  $\phi_a$  denotes the mass fraction of species  $(Y_i)$  or the enthalpy of the mixture (h), and 188  $\phi^{(n)} = \left[Y_1^{(n)}, Y_2^{(n)}, \dots Y_i^{(n)}, h^{(n)}\right]$ . The first term on the right hand side (r.h.s.) represents the 189 convective term while the second term,  $\overline{\rho}S_{\alpha}^{\ r}(\phi^{(n)})dt$ , is due to the chemical reactions. The third 190 term,  $\overline{\rho}S_{\alpha}^{\ s}(\phi^{(n)})dt$ , is the source term due to the spray evaporation. This is different for each 191 species, *i.e.*, solely the evaporating species is involved but it is identical for each stochastic field.

The fourth term corresponds to a gradient transport model for turbulent velocity fluctuation, 192 where  $\Gamma_t = \mu_t / \sigma_t$  is the turbulent diffusivity. Here,  $\mu_t$  is the turbulent viscosity while  $\sigma_t$  is the 193 turbulent Schmidt number  $(Sc_t)$  in the transport equations for chemical species or the Prandtl 194 number ( $Pr_t$ ) in the enthalpy transport equation. Both  $Sc_t$  and  $Pr_t$  are set to 0.7 in the current 195 work. The fifth term, which involves the mixing constant,  $C_{\phi}$ , represents the molecular mixing. 196 The mixing constant is fixed at  $C_{\phi} = 2$ . It is modelled using the Interaction with Exchange to the 197 Mean model.  $\omega_t$  therein is the turbulence frequency obtained from  $\omega_t = \varepsilon / k$  where k and  $\varepsilon$  are 198 199 the turbulent kinetic energy and its dissipation rate, respectively. In the last term on the r.h.s.,  $dW^{(n)}$  represents a vector Wiener process that is spatially uniform but different for each field. 200 The purpose of this term is to introduce a stochastic noise in the transport equations and generate 201 consequent PDF for chemical species and enthalpy. In these simulations,  $dW^{(n)}$  is represented by 202 a time-step increment  $\Delta t^{1/2} \eta_i^n$ , where  $\eta_i^n = \{-1, 1\}$  is a dischotomic random vector. Previous work 203 has shown that the use of thirty-two stochastic fields reached result convergence [35]. The 204 Chemistry Coordinate Mapping (CCM) approach is coupled with the ESF solver in order to 205 integrate the chemical reaction source terms efficiently [36]. In the current work, a four-206 dimensional phase space based on temperature, local equivalence ratio, scalar dissipation rate, 207 208and the mass fraction of fuel is used. Their resolutions are fixed at 5 K, 0.01, 0.025 and 0.001 respectively. The ESF-CCM solver is incorporated with a revised multi-step soot model [14,20]. 209 Mean molar concentrations of acetylene (C2H2) and O2/OH are used to estimate the soot 210 formation and oxidation rates respectively. Both soot and gas radiation heat transfers are not 211 considered in the current work. Bolla et al. [37] showed that the effect of radiation heat transfer 212 influences the local flame temperature in the order of O(10 K) and hence has minor influence on 213 214 local NO and soot concentrations under the Spray A condition. While the effects of radiation heat transfer are more significant at higher  $P_{am}$  due to the increased concentrations of soot, CO<sub>2</sub> and 215 H<sub>2</sub>O, they may only vary the absolute values. Comparisons to the experimental results in section 216

4 show variation of combustion and emissions characteristics with respect to the change of  $P_{am}$ levels are captured by the current model. These qualitative trends are expected to remain unchanged with the consideration of radiation heat transfer.

The computational domain is a constant volume chamber, which has a cubic shape with 220 side lengths of 108 mm. The injector is placed at the center of one of the chamber walls. A 221 uniform Cartesian grid is used. In the previous study, an isotropic cell size of 0.5 mm within the 222 spray combustion region was found to reach mesh independence [35]. This mesh resolution is set 223 as the baseline configuration and mesh sensitivity studies are carried out for different  $P_{am}$  in the 224 current work. The computational grid with the fine, intermediate (baseline) and coarse resolutions 225 consists of approximately 1,440,000, 360,000 and 89,000 cells, respectively. Figure 3 shows that 226 227 the penetration lengths and mixture fraction generated by the finer resolution and the baseline are close while those by the coarser resolution deviate, particularly for liquid fuel penetration length. 228 It should also be mentioned that the qualitative change of the ignition behavior (see Section 4) 229 with respect to the variation of  $P_{am}$  is also consistent when a finer resolution is used. 230

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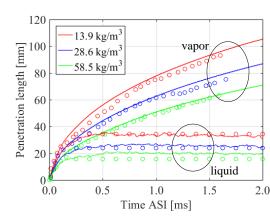
Fig 3.(a) Penetration lengths and (b) mixture fraction distribution at axial position (*x*) near the end of liquid penetration length predicted using different mesh resolution. Dashed-, solid- and dotted-lines are results of the coarse, intermediate (baseline) and fine resolutions, respectively.

236

#### 238 **4. Results and discussion**

#### 239 4.1 Model validation

Figure 4 depicts the temporal evolution of the liquid and vapor penetration lengths of the evaporating, non-reacting sprays from the experiments and the numerical models. The liquid penetration reaches steady state after 0.5 ms, while the vapor penetration continues to grow with time. The model is capable to replicate the experimental penetration lengths reasonably well. As the  $P_{am}$  increases and hence the  $\rho_{am}$  increases, both the liquid and vapor penetration lengths decrease due to the increased droplet drag caused by the higher  $\rho_{am}$ .



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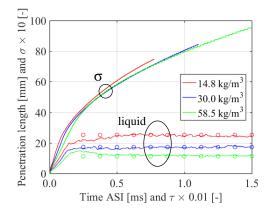
Fig. 4. Liquid and vapor penetration lengths of the evaporating, non-reacting sprays (cases 1 to 3) as a function of time after start of injection (ASI). Solid lines denote the simulation results while markers are obtained by liquid scaling law (liquid) [8] and from the experiment (vapor) [9].

251

Model evaluation is performed further for evaporating, non-reacting sprays with  $D_{nozz}$  of 252 180  $\mu$ m. Since experimental data are not available for  $D_{nozz}$  of 180  $\mu$ m, model evaluation is 253 carried out based on liquid scaling law [8] as well as dimensionless penetration length and time 254 [9]. The current spray breakup model configuration is found applicable for both  $D_{nozz}$  of 180 and 255 257 µm, *i.e.*, the same spray sub-model constants were used. Figure 5 shows that the predicted 256 liquid penetration length of the non-reacting sprays for  $D_{nozz}$  of 180 µm (cases 4 to 6 with  $O_{2,am}$  = 257 0%) agree with the liquid scaling law [4]. The associated vapor penetration length is evaluated 258 using the dimensionless spray tip penetration distance ( $\sigma$ ) and dimensionless penetration time ( $\tau$ ). 259

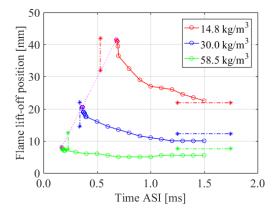
As seen in Fig. 5, all the  $\sigma$  curves collapse, agreeing well with the trend reported in [5]. This indicates that the dependence of vapor penetration length on  $\rho_{am}$  is successfully simulated by the model.

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#### 264

Fig. 5. Predictions of liquid penetration lengths and dimensionless penetration distance (σ) as a
function of dimensionless penetration time (τ). Circle markers denote liquid scaling law results.



268

Fig. 6. Flame lift-off positions of the reacting sprays (cases 4 to 6) as a function of time after start of injection (ASI). Vertical and horizontal dash-dotted lines with asterisks represent the experimental ignition delay times and averaged flame liftoff lengths respectively [3,10]. Magenta dotted line indicates the boundary separating the domain of ignition (larger time ASI) and induction to ignition (smaller time ASI).

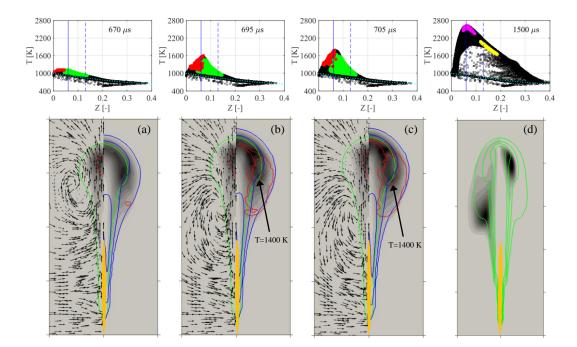
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The decrease in both liquid and vapor penetration lengths lead to the ignition site and the flame stabilization position to move upstream towards the injection tip, reducing the LOL (see Fig. 6). A comparison to the measurements shows that the IDTs and flame LOLs are predicted within 28% for all three cases. The difference is mainly attributed to the overestimated IDT in the 42 bar case. It is interesting to note that a similar simulation result was reported by Bolla et al. [17], when a different TCI closure approach, the conditional moment closure model, was employed with the same mechanism for the same thermochemical conditions. Although the IDTs and LOLs predicted are to a certain extent different from the experiment, the trend of IDT and LOL with respect to the change of  $P_{am}$  is correctly predicted.

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#### 4.2 Autoignition characteristics of diesel spray combustion

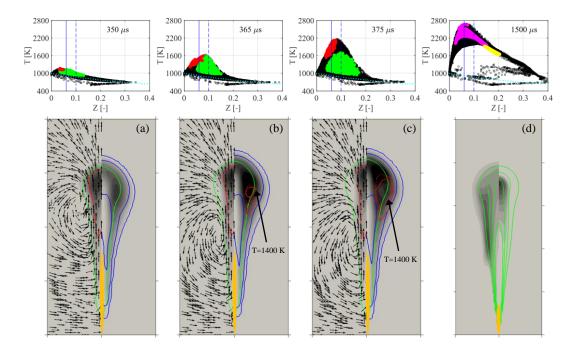
Figures 7 to 9 illustrate the autoignition characteristics of the three cases using scatter plots of 286 temperature (T)-mixture fraction (Z) and contours of various combustion products. In the 42 bar 287 case, RO<sub>2</sub> is formed in the fuel-rich inner core of the spray jet with a temperature around 750 K, 288starting at approximately 350 µs (not shown). Through the low-temperature reaction pathway, 289 290 RO<sub>2</sub> oxidizes to ketohydroperoxide. The HO<sub>2</sub> radical is formed in the vicinity of ketohydroperoxide, extending to the fuel-lean side of the stoichiometric line. These species are 291 formed downstream of the liquid fuel. The first-stage ignition is found across a wide range of 292 mixture fraction. As shown in Fig. 7(a), a small amount of OH is formed at 670 µs on the fuel-293 lean side and the peak temperature increases above 1100 K. Figure 7(b) depicts that HO<sub>2</sub> reaches 294 a higher concentration in the fuel-rich zone at 695 µs, as compared to that in the fuel-lean zone. 295 HO<sub>2</sub> in this region is first consumed and OH is formed. The local temperature increases beyond 296 the temperature threshold of 1400 K (400 K above  $T_{am}$ ), indicating the onset of the main ignition. 297 Rate of production (ROP) analysis of the 0-D simulation of  $Z_{mr}$  (in section 4.1) shows that OH is 298 mainly formed from the sequence HO<sub>2</sub>+HO<sub>2</sub> $\leftrightarrow$ H<sub>2</sub>O<sub>2</sub>+O<sub>2</sub> (R1), H<sub>2</sub>O<sub>2</sub>+M $\leftrightarrow$ 2OH+M (R2), where M299 is the third body species. It is also noticed that the main ignition occurs in the region where the 300 scalar dissipation rate ( $\gamma$ ) is low and the residence time is long (within the center of the 301 recirculation zone). 302



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Fig. 7. Scatter plots of temperature-mixture fraction (upper row) and contours of combustion 304 products (bottom row) in the 14.8 kg/m<sup>3</sup> case a) at first-stage ignition, b) at onset of second-stage 305 ignition, c) towards the formation of a classical diesel spray flame and d) during quasi-steady 306 state. Upper row: Cells with substantial formation of HO<sub>2</sub>, OH, NO and soot volume fraction 307  $(f_{v,soot})$  are marked with green, red, magenta and yellow respectively. The thresholds are set to 308 309 40% of their respective maximum mass fraction. Stoichiometric lines, the most reactive mixture 310 fraction and mixing lines are represented by solid blue, dashed blue and dotted cyan lines respectively. Bottom row (a)-(c): (Left) Ketohydroperoxide contour with flowfield (black 311 arrows),  $RO_2$  of 1 x 10<sup>-4</sup> (solid red); and (right) HO<sub>2</sub> contour with isolines of scalar dissipation 312 rates of 1, 10 and 100 (solid blue), OH mass fraction of 1 x 10<sup>-4</sup> (solid red), temperature of 1400 313 K (dotted red); (d) (Left) NO and (right)  $f_{v,soot}$  contours at quasi-steady state with isolines of local 314 equivalence ratio of 1 to 4 (solid green). Maximum mass fraction of ketohydroperoxide, HO<sub>2</sub>, and 315 NO is fixed at 0.03, 0.001 and 0.002 respectively while maximum soot volume fraction is set to 5 316 ppm. Solid green lines indicate the stoichiometric line and the orange lines represent the averaged 317 liquid length. For (a) to (c), each frame shows 30 x 60 mm while for (d), the frame shows 40 x 318 100 mm. 319

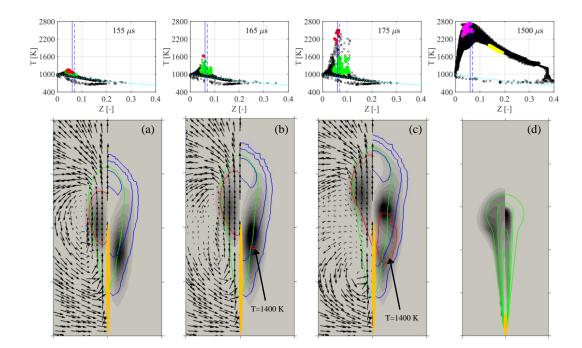
The formation of RO<sub>2</sub>, ketohydroperoxide and HO<sub>2</sub> starts earlier at approximately 200 µs 321 in the 85 bar case (not shown). Figure 8(a) depicts that OH formation is formed at 350 µs in the 322 fuel-lean region where the local temperature exceeds 1100 K. The second-stage ignition then 323 takes place at 365  $\mu$ s on the fuel-rich side of the stoichiometric line, in the region where  $\gamma$  is low 324 and the residence time is long. These phenomena are akin to those of 42 bar case discussed 325 above, although the transition from low- to high-temperature ignition has a shorter duration. A 326 shorter transition is also observed in the homogenous reactor calculation (cf. Fig. 2). Reaction 327 path analysis shows that reaction R2 remains the most significant step during the main ignition 328 event. When  $P_{am}$  increases, the separation between the end of liquid fuel and key radicals such as 329 RO<sub>2</sub>, ketohydroperoxide and HO<sub>2</sub> reduces. The ignition site is closer to the penetration tip where 330 the local Z value is higher. In the 42 and 85 bar cases, the mixture on the fuel-lean side first 331 experiences a higher temperature rise, presumably due to the higher  $T_{am}$ . It then promotes the 332 low-temperature reactions in the rich mixture due to the transport of warmer lean products into 333 334 the fuel-rich zone. These phenomena are similar to those reported in Refs. [4-7], which studied the ignition behavior for the Spray A condition ( $O_{2,am} = 15\%$ ;  $T_{am} = 900$ K;  $P_{am} = 60$  bar). The 335 observation that the main ignition event occurs in more fuel-rich mixtures in the case with higher 336  $P_{am}$  is consistent with the findings of Higgins et al. [12] who experimentally studied the ignition 337 behavior at different  $\rho_{am}$  in the same combustion vessel. However, the behavior changes when 338  $P_{am}$  is further increased. As discussed next, the current results show that the mixture fraction of 339 the first igniting mixture during the high-temperature ignition does not vary monotonically with 340 *P<sub>am</sub>* and the second-stage ignition takes place in a more fuel-lean region in the 170 bar case. 341



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Fig. 8. Scatter plots of temperature-mixture fraction (upper row) and contours of combustion products (bottom row) in the 30.0 kg/m<sup>3</sup> case at different times. Descriptions can be found in the caption of Fig. 7, apart from maximum soot volume fraction which is set to 50 ppm. For (a) to (c), each frame shows 20 x 40 mm while for (d), the frame shows 40 x 100 mm.

At 170 bar, RO<sub>2</sub>, ketohydroperoxide and HO<sub>2</sub> are formed early at 100 µs (not shown). The 349 temperature of the fuel-lean mixture increases to above 1100 K at 150 µs. Under this condition, 350 the transition from low- to high-temperature ignition is even faster, occurring in less than 15 µs. 351 352 The transport of warmer products from the fuel-lean zone into the fuel-rich mixture is relatively slow, as compared to the increased reaction rate. Figure 9 depicts that there is no separation in the 353 spray direction between HO<sub>2</sub> and liquid fuel, *i.e.*, HO<sub>2</sub> is formed in the vicinity of the fuel spray 354 region. Within the fuel-rich region, the local temperature is low due to evaporation and the 355 condition is not appropriate for ignition. The scatter plots in Figs. 8(b) and 9(b) support that, 356 during the high-temperature ignition, the local temperature within the fuel-rich region becomes 357 lower when P<sub>am</sub> increases. Eventually, the high-temperature ignition occurs on the fuel-lean side 358 of the stoichiometric line. 359



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Fig. 9. Scatter plots of temperature-mixture fraction (upper row) and contours of combustion products (bottom row) in the 58.5 kg/m<sup>3</sup> case at different times. Descriptions can be found in the caption of Fig. 3, apart from maximum soot volume fraction which is set to 500 ppm. For (a) to (c), each frame shows  $5 \times 20$  mm while for (d), the frame shows  $40 \times 100$  mm.

The ignition characteristics from low- to high-temperature ignition in the three cases are also 367 illustrated using the scatter plots of rate of fuel concentration change and mixture fraction. These 368 plots can be found in Fig. 10. A positive value indicates the presence of vapor fuel while a 369 negative value denotes the fuel consumption. Each plot is colored by the normalized local 370 temperature at that particular timestep, *i.e.*, dark blue, blue, green and yellow represent low, 371 intermediate, high, the highest temperature, respectively. At 42 and 85 bar, the higher 372 temperature regime first falls on the fuel-lean mixture and then shifts to the fuel-rich mixture. 373 During the main ignition process, the fuel has been consumed where the main ignition event 374 occurs. On the other hand, at 170 bar, the vapor fuel is still observed but the corresponding 375 mixture is cold and does not get involved in the ignition process. The mixture on the fuel-lean 376 side of the stoichiometric line undergoes low- and then high-temperature ignition. These support 377 the discussion earlier based on Figs 7 to 9. 378

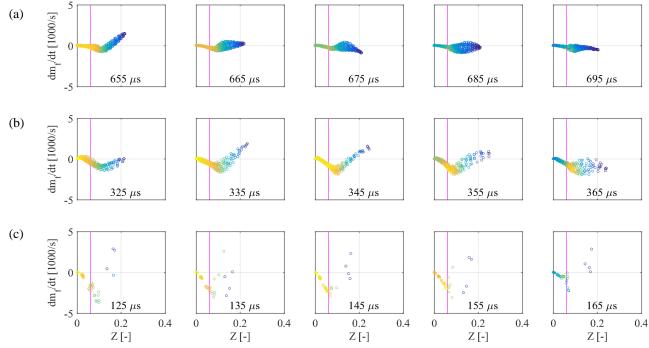
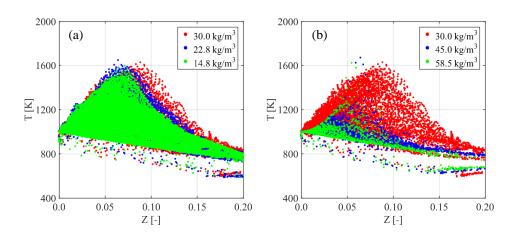


Fig. 10. Scatter plots of rate of fuel concentration change and mixture fraction for (a) the 14.8 kg/m<sup>3</sup> case at 44 mm from the injector tip, (b) the 30.0 kg/m<sup>3</sup> case at 26 mm from the injector tip and (c) the 58.5 kg/m<sup>3</sup> case at 8 mm from the injector tip. The last column provides scatter plots at their respective main high temperature ignition.

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Fig. 11. Scatter plots of temperature-mixture fraction of (a) the 14.8 kg/m<sup>3</sup> case at 695  $\mu$ s, the 22.8 kg/m<sup>3</sup> case at 470  $\mu$ s, the 30.0 kg/m<sup>3</sup> case at 365  $\mu$ s and (b) the 30.0 kg/m<sup>3</sup> case at 365  $\mu$ s, the 45.0 kg/m<sup>3</sup> case at 230  $\mu$ s and the 58.5 kg/m<sup>3</sup> case at 165  $\mu$ s.

Computation of two additional intermediate cases of  $\rho_{am}$  of 22.8 kg/m<sup>3</sup> (67 bar) and 45.0 391  $kg/m^3$  (128 bar) confirms this trend. Figure 11 demonstrates that the main ignition occurs in more 392 fuel-rich mixtures when  $P_{am}$  increases from 42 bar to 85 bar, consistent with results of Higgins et 393 al. [12]. Above 85 bar, the mixture that undergoes the main ignition becomes more fuel-lean 394 when  $P_{am}$  increases. In order to further verify this trend, simulations are performed using the 395 Lu68 mechanism [16] for cases 5 and 6. The predicted IDT using the Lu68 mechanism is slightly 396 closer to the measurement in the 170 bar case (case 6), with a relative difference of < 8% from 397 the measurement, as compared to that using the Liu44 mechanism with a relative difference of 398 ~10% from the measurement. The high-temperature ignition occurs on the fuel-rich side (Z of 399 0.109) and the fuel-lean side (Z of 0.0603) in the 85 bar (case 5) and 170 bar (case 6) 400 respectively, showing that the qualitative change of the ignition behavior with respect to the 401 variation of Pam from 85 bar to 170 bar is the same for the Liu44 and Lu68 mechanisms. It should 402 be noted that the variation of Z of the first ignition mixtures during the high-temperature ignition 403 404 is identified for these particular injection specifications, initial ambient gas composition and  $T_{am}$ . As the ignition behavior is dependent of both physical and chemical processes, the variation of Z405 of the first ignition mixtures during the high-temperature ignition with respect to  $P_{am}$  may 406 change. Understanding the ignition behavior is crucial for development of skeletal surrogate fuel 407 mechanisms. Once the ignition behavior is identified for a narrower range of operating 408 conditions, the size of the skeletal surrogate fuel mechanisms can be further optimized to increase 409 the computational efficiency. 410

Effects of  $P_{am}$  on the quasi-steady state flame structures will be discussed next. Upon the ignition point, it takes approximately 160 hours for 0.1 ms on ten 64-bit Intel Ivy Bridge Xeon E5-2650 v2 8-core CPUs running at 2.60 GHz with the Liu44 mechanism. The computational cost is expected to escalate significantly with the use of the Lu68 mechanism. Hence, only the Liu44 mechanism is used for the simulations up to quasi-steady state and the associated results are used for the flame structure analyses.

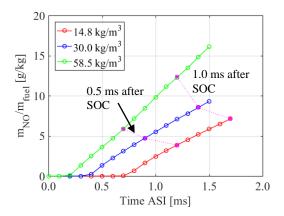


Fig. 12. Normalized total mass of NO as a function of time after start of injection (ASI). Magenta
dotted lines indicate the total NO mass production remains higher after their respective start of
combustion (SOC).

An analysis based on transport budgets [38] near the lift-off position at quasi-steady state suggests the flame is stabilized by the auto-ignition process. This remains unchanged for three  $P_{am}$  levels and agrees with the findings from the literature. In addition to that, the Takeno's flame index (FI) is used to investigate the spray flame structure. The local equivalence ratio is incorporated into FI such that the premixed flame can be identified for both fuel-lean and fuelrich conditions, as shown in Eq. (2),

428 
$$FI = \frac{\nabla Y_F \cdot \nabla Y_O}{\left|\nabla Y_F \cdot \nabla Y_O\right|} \cdot \frac{\phi - 1}{\left|\phi - 1\right|}$$
(2)

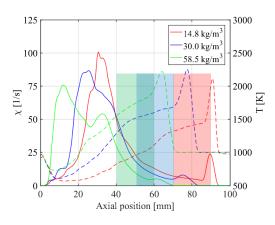
where the mass fraction of fuel,  $Y_F$  includes those of evaporating species, *n*-heptane as well as the 429 main fuel-rich combustion products, *i.e.* CO and H<sub>2</sub> [35]. The quasi-steady spray structures are 430 found independent from the effects of  $P_{am}$  at the tested initial  $T_{am}$  and ambient gas composition. 431 Takeno's FI shows that all the spray flames have the classical structure of the diesel spray flames 432 described in the Dec conceptual model, *i.e.*, a premixed rich flame upstream embedded by a 433 diffusion flame which extends downstream [40]. Figures 7(d), 8(d) and 9(d) demonstrate that the 434 location of NO and soot formation regions for the three  $P_{am}$  levels. High NO concentration is 435 found above 2200 K near the stoichiometric mixture, while high soot concentration is in the 436 premixed fuel-rich zone with local temperature ranging from 1600 to 2200 K; this result is 437

consistent with the literature [40,41]. A decrease in the flame size (in terms of both flame length 438 and width) with the increase of  $P_{am}$ , can also be seen in Figs. 7(d), 8(d) and 9(d). An evaluation 439 of the reaction zone thickness based on the full-width at half maximum of OH mass fraction at 440 the middle of the spray flames shows that the thickness of the reaction zone is approximately 7.2, 441 3.2 and 1.6 mm for the 14.8, 30.0 and 58.5 kg/m<sup>3</sup> cases, respectively. These trends are in 442 qualitative agreement with the experimental images based on OH chemiluminescence [11]. The 443 reaction zone thickness for the O radicals exhibits a similar trend as those of OH. The spatial NO 444 distribution becomes narrower when  $P_{am}$  increases. Although the reaction zone size decreases 445 with the increasing ambient pressure, the rate of NO formation increases more rapidly with 446 increasing  $P_{am}$ . The increased reaction rate attributed by the increased  $P_{am}$  is more influential to 447 the final NO production. The total NO formation hence increases with  $P_{am}$ , as indicated by the 448 magenta dotted lines in Fig. 12. The current model predicts that the peak soot volume fraction 449  $(f_{v,soot})$  increases approximately six-fold when  $P_{am}$  increases from 42 to 85 bar and further 450 451 increases approximately thirteen-fold when  $P_{am}$  further increases from 85 to 170 bar. To date, the revised multi-step model has only been evaluated using  $f_{v,soot}$  measurements collected at 42 and 452 85 bar [14,20] since  $f_{v,soot}$  measurement above 85 bar is not available. Further work is thus 453 required to confirm the estimation of  $f_{v,soot}$  at 170 bar. Yet, the trend of maximum  $f_{v,soot}$  increasing 454 with  $P_{am}$  agrees with experimental observations [13]. 455

Figure 2 shown earlier illustrates that the ignition site is placed further upstream and the 456 flame stabilizes more quickly when  $P_{am}$  increases. At quasi-steady state, the flame LOL reduces 457 with increasing  $P_{am}$ . This reduces the air entrainment, giving rise to a higher mixture fraction 458 within the fuel-rich core region. Besides that, the elevated reaction rate at higher  $P_{am}$  also 459 promotes soot formation. These factors are known to increase the local peak  $f_{v,soot}$  [13]. The 460 current study shows that the rise of the soot formation rates can be attributed to other factors. As 461 aforementioned, the flame width reduces for a higher  $P_{am}$ . Consequently, the stoichiometric line 462 or the mixing-controlled combustion layer is closer to the central region with higher  $P_{am}$ , which 463

increases the heat transfer. As shown in Fig. 13, the temperature in the rich premixed region increases with  $P_{am}$ , contributing to the higher soot formation rates. Apart from these, it is noticed that both the velocity near the spray region and  $\chi$  decrease with the rise of  $P_{am}$ . This indicates the mixing rate within the fuel-rich region decreases, promoting soot precursor and particle formation within the fuel-rich region. These collectively contribute to the higher local  $f_{v,soot}$  when  $P_{am}$ increases.

470





472 Fig. 13. Scalar dissipation rate,  $\chi$  (solid lines) and temperature (dashed lines) along the spray axis. 473 Semi-transparent windows indicate the domains with high soot volume fraction as can be seen in 474 Figs. 7 to 9.

475

#### 476 Concluding remarks

The contribution of this study is the generalization of the ignition behaviors and flame structures 477 at different ambient pressures  $(P_{am})$  under diesel engine-like conditions. Diesel #2 spray flames at 478 three P<sub>am</sub> levels of 42, 85 and 170 bar have been studied using an Eulerian-based transported 479 probability density function model in three-dimensional computational fluid dynamic 480 simulations. Comparisons to measurements show that the ignition delay times and flame lift-off 481 lengths are predicted fairly well. The numerical results show that the mixture fraction of the first 482 igniting mixture during the high-temperature ignition does not vary monotonically with  $P_{am}$ . The 483 high-temperature ignition takes place on the fuel-rich side in the 42 bar and 85 bar cases. 484 However, at 170 bar the high-temperature ignition occurs on the fuel-lean side due to the 485

increased reaction rates and temperature in the fuel-lean mixture. The main ignition in all three 486 cases occurs in the region where the scalar dissipation rate ( $\gamma$ ) is low and the residence time is 487 long. Analyses of the quasi-steady spray flame structures reveal that the stoichiometric mixtures 488 are closer to the fuel-rich core when  $P_{am}$  is increased, leading to higher heat transfer and 489 increased temperature within the fuel-rich core. Besides that,  $\chi$  reduces in cases with higher  $P_{am}$ . 490 These collectively contribute to the higher local soot volume fraction when  $P_{am}$  increases, in 491 addition to poorer air entrainment attributed by the reduced lift-off length and the increased 492 reaction rates. 493

494

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