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1 **Dynamic simulation of Adiabatic Compressed Air Energy Storage (A-CAES) plant with integrated**
2 **thermal storage – link between components performance and plant performance**

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12
13 **Abstract**

14 The transition from fossil fuels to green renewable resources presents a key challenge: most renewables are
15 intermittent and unpredictable in their nature. Energy storage has the potential to meet this challenge and
16 enables large scale implementation of renewables. In this paper we investigated the dynamic performance of
17 a specific Adiabatic Compressed Air Energy Storage (A-CAES) plant with packed bed thermal energy storage
18 (TES). We developed for the first time a plant model that blends together algebraic and differential sub-models
19 detailing the transient features of the thermal storage, the cavern, and the compression/expansion stages. The
20 model allows us to link the performance of the components, in particular those of the thermal storage system,
21 with the performance of the whole A-CAES plant. Our results indicate that an A-CAES efficiency in the range
22 60-70% is achievable when the TES system operates with a storage efficiency above 90%. Moreover, we show
23 how the TES dynamic behaviour induces off-design conditions in the other components of the A-CAES plant.
24 Such device-to-plant link of performance is crucial: only through integration of TES model in the whole A-
25 CAES model is possible to assess the benefits and added value of thermal energy storage. To the authors'
26 knowledge the present study is the first of this kind for an A-CAES plant.

Nomenclature

A	Area (m ²)
C	Heat capacity rate (J s ⁻¹ K ⁻¹)
D, d	Diameter (m)
c_p	Specific heat (J kg ⁻¹ K ⁻¹)
Ex_{in}	Exergy flux (W K ⁻¹)
\dot{G}	Reduced flow rate (-)
h	Specific enthalpy (J kg ⁻¹)
h_v	Volumetric heat transfer coefficient (W m ⁻³ K ⁻¹)
H	Height (m)
k	Specific heat ratio (-)
k_a, k_s	Thermal conductivity (W m ⁻¹ K ⁻¹)
\dot{m}	Mass flow rate (kg s ⁻¹)
m	Mass (kg)
\dot{n}	Reduced speed (-)
p	Pressure (Pa)
T	Temperature (K)
U	Overall heat transfer coefficient (W m ⁻² K ⁻¹)
u	Velocity (m s ⁻¹)
W	Power (W)

Greek letters

α	Influence factor (-)
β	Compression ratio (-)
ε	Effectiveness, void fraction (-)
η	Isoentropic efficiency (-)

η_{cycle}	Round trip efficiency (-)
η_{th}	Thermal storage efficiency (-)
π	Expansion ratio (-)
ρ	Density (kg m^{-3})
Φ	Heat transfer rate (W)

27

28

29 **1 Introduction**

30 In 2013 the electricity production has reached 23 000 TWh/year of which oil, natural gas, and other fossil fuels
31 account for 68% while renewable sources contribute for less than 6% [1]. Overcome this energy scenario is
32 imperative as CO₂ emissions and global warming are already taking their toll on our society and planet Earth
33 [2]. To contain global warming below 2°C carbon dioxide emission must decrease by 90% by 2050 through
34 an intense penetration of renewable resources which could reach a global share of 65% according to scenarios
35 forecasted by IEA [3]. This great potential can be untapped only if the intrinsic variability of renewables, such
36 wind and solar energy, is mitigated through energy storage (ES). ES technology provides several functions to
37 facilitate the use of renewables: it enables to capture “wrong time” energy and make it available when needed,
38 it helps to shave and shift load peaks, and it improves reliability of energy systems [4,5].

39 Alongside with pumped hydroelectricity storage, compressed air energy storage (CAES) is among the few
40 grid-scale energy storage technology with power rating of 100s MW [6,7]. CAES operates in such a way that
41 electrical energy is stored in the form of compressed air confined in a natural or artificial reservoir. Then,
42 during periods of high energy demand, stored energy is retrieved by withdrawing high pressure air and expand
43 it through a series of turbines to generate electricity. Traditionally, for example in the Huntorf plant [7,8],
44 before expansion air is heated in a combustion chamber burning conventional fossil fuels. This leads to two
45 drawbacks: CAES is not CO₂ free and round trip efficiency is limited to 40-50% [6,7]. To overcome such
46 disadvantages adiabatic compressed air energy storage (A-CAES) has been proposed. Instead of burning fuel,
47 in A-CAES the heat generated by compression is stored in a Thermal Energy Storage (TES) and then used to
48 heat air from the reservoir before it enters the turbines [7,9]. As a result, round trip efficiency increases to 70-
49 75% according to [7,10,11] and fuel consumption is avoided. The vast majority of the studies on A-CAES
50 consider indirect heat exchangers (HEXs) and a separate thermo-fluid to store the heat of compression [9,11-
51 18]. The heat of compression, exchanged via air-to-fluid HEXs, increases the internal energy of the working
52 fluid which acts as a sensible heat storage medium. Commonly, HEXs have been considered installed between
53 each compression stage, to store heat, and between each expansion stage to retrieve heat during discharge fo
54 ACAES plant [11-18].

55 Another proposed A-CAES configuration uses a solid medium, typically natural rocks, to store the heat of
56 compression [7,19]: during A-CAES charging heat is stored by flowing hot air from compressors through a
57 packed bed of rocks; when discharge occurs air from the cavern flows through the packed bed retrieving the
58 heat previously stored and then expands through turbines train to generate electricity. Literature presents
59 multiple studies on packed beds dealing with the design [20-24], the heat transfer performance [25-30], and
60 the effect of operating conditions [31-34]. However, the *dynamic performance* of A-CAES plant with an
61 integrated packed bed thermal storage remain unaddressed. With this study we fill such a gap in the literature
62 by presenting for the first time a full investigation of an A-CAES plant with packed bed thermal storage. The
63 mathematical model we developed is fully dynamic and it includes off-design performance of each component
64 of the A-CAES plant. The model blends together algebraic and differential sub-models that detail the transient
65 features of the thermal storage, the cavern, and the compression/expansion stages. This allows to link the
66 performance of the components, in particular those of the thermal storage system, with the performance of the

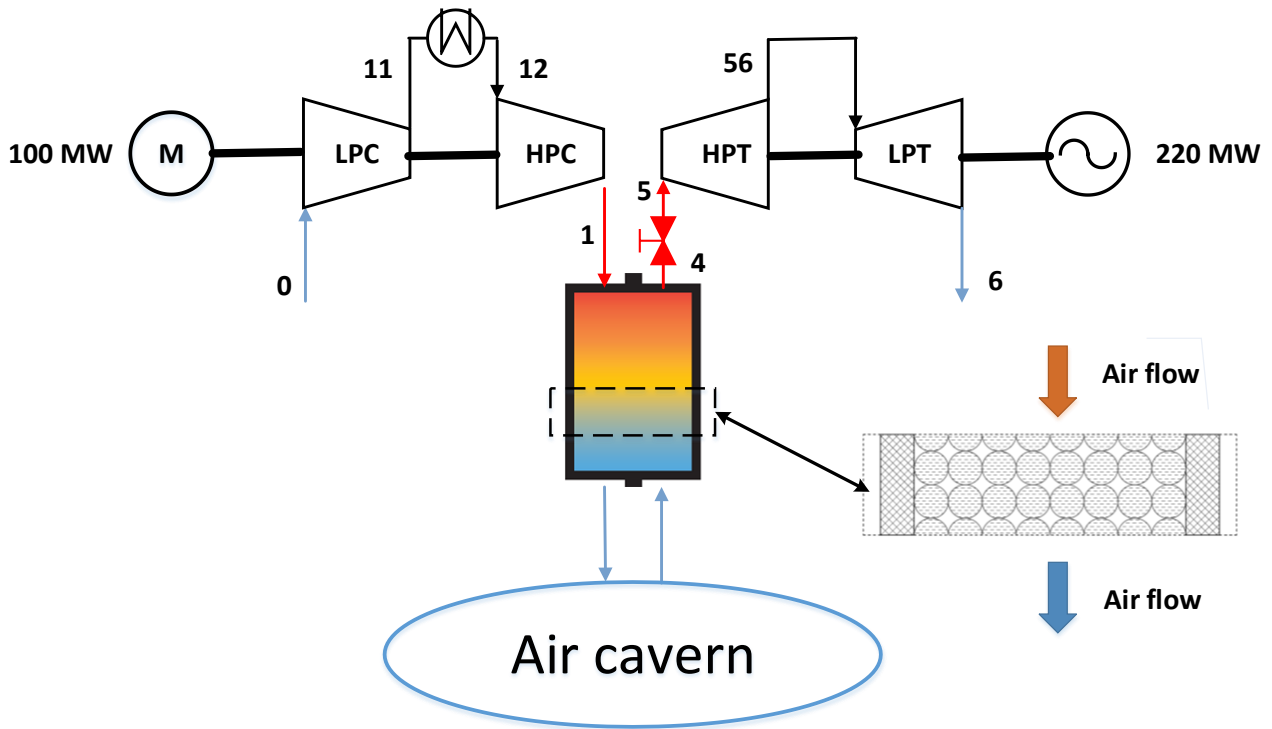
67 whole A-CAES plant. Such device-to-plant link is crucial: only through integration of TES in the whole A-
68 CAES system is possible to assess the benefit and added value of thermal energy storage. To the authors’
69 knowledge the present study is the first of this kind for an A-CAES plant.

70 **2 System description**

71 Figure 1 presents the specific adiabatic compressed air energy storage system (A-CAES) studied in this work.
72 Table 1 summarizes the major features of the A-CAES plant. A packed bed thermal energy storage (TES)
73 ensures the “adiabatic” conditions: after the HPC compression stage, hot air flows through the packed bed and
74 exchanges heat with the gravel contained in the TES. The gravel acts as sensible storage material and captures
75 heat for later purposes. Air leaves the TES system nearly at ambient temperature and enters the cavern at high
76 pressure. It is worth noting that we focused on a specific A-CAES configuration. A Similar plant configuration
77 is also considered by RWE Power in the EU project “ADELE” [35] and by Airlight Energy [36] although other
78 A-CAES designs are also possible [6,11,14,17,18].

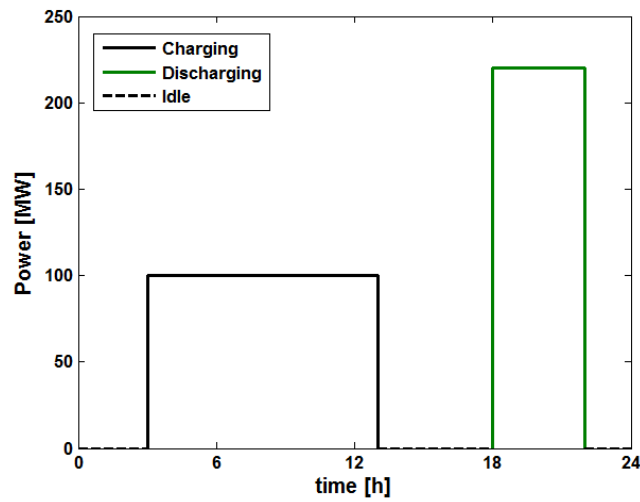
79 An inter-refrigeration heat exchanger cools the air flow before it enters the high pressure stage. This
80 configuration was also considered in [7] to prevent excessively high air temperature at the outlet of HPC. The
81 compressors operate over a range of compression ratios since air pressure in the cavern spans the range 46 to
82 72 bar, which is the typical range adopted for the Huntorf plant and Machintosh plant [7,8]. The cavern’s size
83 considered is a typical one for natural salt caverns [6,7]. During the discharge process, energy is retrieved by
84 withdrawing air from the cavern at high pressure and expand it through the train of turbines. Two discharge
85 modes have been considered in the literature: variable inlet pressure and constant inlet pressure [7,37]. In the
86 former one high pressure air from the cavern directly expands through the turbines which therefore experience
87 a variable (in time) expansion ratio. We considered constant inlet pressure mode: as depicted in Fig. 1, a
88 throttling system maintains the turbine inlet pressure constant. Such an operating mode allows to operate the
89 turbine train at constant expansion ratio and near to design conditions – thus at maximum efficiency – for the
90 entire discharge process. Design expansion ratio (Table 1) for HPT and LPT were chosen as the one for existing
91 CAES plants [8]. Tables 2 and 3 present the thermodynamic state points for compression and expansion under
92 design conditions. The thermodynamic properties were evaluated with EES (Engineering equation solver)
93 using the data in Table 1 as input parameter. For the design conditions reported in Tables 2 and 3 we considered
94 the same temperature for the air temperature at compressor outlet (point 1) and the air temperature at the outlet
95 of TES (point 4). Clearly, a temperature drop is expected under operation (that is $T_1 > T_4$) because of finite
96 heat transfer between air and the filling material of the TES. As illustrated in the Results section T_1 and T_4
97 differs minimally which support the assumption, for design calculations, of $T_1 = T_4$.

98 For the purpose of simulation of A-CAES plant operation we considered n equal cycles of 10 hours charge, 4
99 hours discharge and 10 hours idle, as shown in Figure 2. Such a figure present the nominal cycle with constant
100 power input during charge and constant power output during discharge. The actual profile of each cycle was
101 determined through the simulations performed, as detailed in the Results section. The nominal profile of Fig.
102 2 was chosen considering the A-CAES plant operating for peak shaving, minute reserve, or compensation of
103 fluctuation in wind power. Such operation modes are typical of existing CAES plants [7,8], and present
104 discharge time of 3-4 hours, as in the case of Fig. 2. The total number n of cycles considered in the study was
105 30.



106

107 Figure 1: Adiabatic compressed air energy storage (A-CAES) plant with sensible thermal energy storage.



108

109 Figure 2: Charge and discharge cycle.

110 Table 1. Major parameters of A-CAES system

Quantity	Value
Ambient temperature	293.15 K
Ambient pressure	1.01325 bar
Expansion train rated power	220 MW
HP turbine design inlet temperature	905.15 K
HP turbine design inlet pressure	46 bar
HP turbine design expansion ratio	4.18
LP turbine design inlet temperature	655.15 K
LP turbine design inlet pressure	11 bar
LP turbine design expansion ratio	11
Turbines design efficiency	88%
Compression train rated power	100 MW

HP compressor design inlet temperature	480.15 K
HP compressor design compression ratio	8.4
LP compressor design compression ratio	8.4
Cavern volume	230 000 m ³
Cavern min/max pressure	46/72 bar
Cavern wall heat transfer coefficient [43]	$0.02356 + 0.0149 \dot{m}_{in} - \dot{m}_{out} ^{0.8}$

111

112

Table 2. Thermodynamic states for the charging process

State	Temperature [°C]	Pressure [bar]	Enthalpy [kJ/kg]	Entropy [kJ/(kg*K)]	Mass flow rate [kg/s]
0	20	1.01325	300.31	6.87	120
11	309.0	8.5413	588.54	6.93	120
12	207.0	8.5413	482.43	6.73	120
1	632.2	72.0	943.12	6.79	120

113

114

Table 3. Thermodynamic states for discharging process

State	Temperature [°C]	Pressure [bar]	Enthalpy [kJ/kg]	Entropy [kJ/(kg*K)]	Mass flow rate [kg/s]
4	632.2	72.0	943.12	6.79	380
5	633.4	46.0	943.12	6.93	380
56	382.3	11.0	666.44	6.98	380
6	134.2	1.01325	408.84	7.17	380

115

116

3 Mathematical modelling of A-CAES plant and validation

117

This section presents the mathematical models for each component of the A-CAES plant depicted in Fig. 1.

118

Each model is first presented separately along with the underlying assumption adopted in the study. The section

119

ends with the description of the solution strategy used to link each sub-model to simulate the whole A-CAES

120

plant. Where not stated explicitly the modelling was performed in Matlab/Simulink 2014 [38].

121

Compressors

122

Modelling of low pressure compressor (LPC) and high pressure compressor (HPC) involves mass and energy

123

balance in order to compute temperature of air exiting each stage and the compression work. Isoentropic air

124

outlet temperature was computed as:

125

$$T_{c,out}^{is} = T_{c,in} (\beta_i)^{\frac{k-1}{k}} \quad (1)$$

126

where $\beta_i = \beta_{HPC}, \beta_{LPC}$ is the compression ratio of each stage. Actual outlet temperature $T_{c,out}$ was obtained using

127

compressor isoentropic efficiency defined as:

128

$$\eta_c = \frac{T_{c,out}^{is} - T_{c,in}}{T_{c,out} - T_{c,in}} \quad (2)$$

129

The power of compressors consumed during charge was evaluate by an energy balance at each compressor

130

neglecting variations in inlet to outlet kinetic energy of air:

131 $W_c = \dot{m}_c (h_{c,out} - h_{c,in})$ (3)

132 In this work we considered off-design performance of compressors during the operation of the A-CAES plant.
 133 Off-design are commonly included in models of energy systems, however CAES systems are often studied
 134 considering only design conditions [9,11,18]. Such an approach may neglect important dynamic effects when
 135 compression train model is included in the whole A-CAES plant model, as we will show in the Results section.
 136 We included off-design calculations through compressors characteristic maps [39] that quantify compression
 137 ratios β_i and isentropic efficiency η_i as function of dimensionless flow rate. The characteristic maps were
 138 approximated according to [40], namely:

139 $\beta_i = c_1 (\dot{G}_c)^2 + c_2 \dot{G}_c + c_3$ (4)

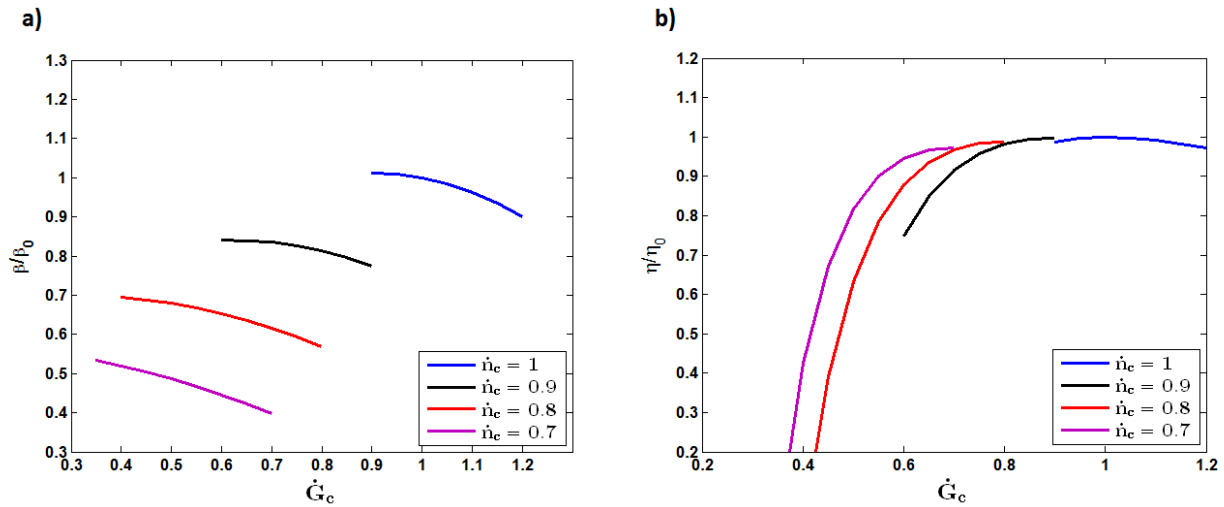
140 $\eta_c = [1 - c_4 (1 - \dot{n}_c)^2] \dot{n}_c / \dot{G}_c (2 - (\dot{n}_c / \dot{G}_c))$ (5)

141 where \dot{G}_c and \dot{n}_c are the reduced flow rate and the reduced speed, respectively. Figure 3 presents the
 142 characteristic maps for the compressors. The definitions for reduced quantities and coefficients of Eqs. 4 and
 143 5 are the following ones:

144 $\dot{G}_c = (\dot{m}_c \sqrt{T_{c,in}} / P_{c,in}) / (\dot{m}_c \sqrt{T_{c,in}} / P_{c,in})_0$
 $\dot{n}_c = (n_c / \sqrt{T_{c,in}}) / (n_c / \sqrt{T_{c,in}})_0$ (6)

145 $c_1 = \dot{n}_c / \left[p \left(1 - \frac{m}{\dot{n}_c} \right) + \dot{n}_c (\dot{n}_c - m)^2 \right]$
 $c_2 = (p - 2m\dot{n}_c^2) / \left[p \left(1 - \frac{m}{\dot{n}_c} \right) + \dot{n}_c (\dot{n}_c - m)^2 \right]$ (7)
 $c_3 = -(pm\dot{n}_c - m^2\dot{n}_c^3) / \left[p \left(1 - \frac{m}{\dot{n}_c} \right) + \dot{n}_c (\dot{n}_c - m)^2 \right]$

146 Subscript 0 in previous equations denotes design conditions while $p = 1.8$, $m = 1.4$ and $c_4 = 0.3$ [40].



147
 148 Figure 3: Characteristic maps for the compressors. a) Compression ratio vs. reduced flow rate; b) Isoentropic
 149 efficiency vs. reduced flow rate.

150 *Turbines*

151 HP and LP turbine were modelled through mass and energy balance following the same approach adopted for
 152 the compressors. Defined the expansion ratio as $\pi_t = p_{in}/p_{out}$, the temperature of air exiting each turbine stage
 153 was obtained from the isentropic temperature and the definition of isentropic efficiency:

$$154 \quad T_{t,out}^{is} = T_{t,in} / \left(\pi_t \right)^{\frac{k-1}{k}} \quad (8)$$

$$155 \quad \eta_t = \frac{T_{t,in} - T_{t,out}}{T_{t,in} - T_{t,out}^{is}} \quad (9)$$

156 where $\pi_i = \pi_{HPT}, \pi_{LPT}$. Finally, the power output was calculated as

$$157 \quad W_t = \dot{m}_t (h_{t,out} - h_{t,in}) \quad (10)$$

158 An improved Flugel formula [40] was used to describe the off-design performance of turbines:

$$159 \quad \frac{\dot{m}_t}{\dot{m}_{t0}} = \alpha \sqrt{\frac{T_{t0,in}}{T_{t,in}}} \sqrt{\frac{\pi_t^2 - 1}{\pi_{t0}^2 - 1}} \quad (11)$$

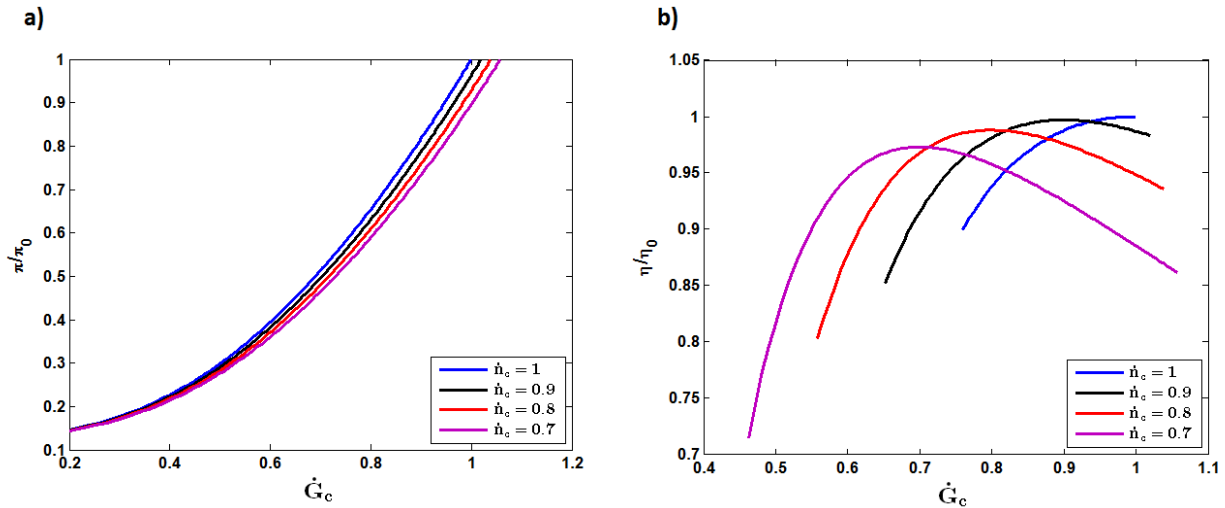
$$160 \quad \frac{\eta_t}{\eta_{t0}} = \left[1 - t(1 - \dot{n}_t)^2 \right] \left(\dot{n}_t / \dot{G}_t \right) \left(2 - (\dot{n}_t / \dot{G}_t) \right) \quad (12)$$

161 The definition for reduced flow and reduced speed for turbines are:

$$162 \quad \dot{G}_t = \left(\dot{m}_t \sqrt{T_{t,in}} / P_{t,in} \right) / \left(\dot{m}_t \sqrt{T_{t,in}} / P_{t,in} \right)_0 \quad (13)$$

$$\dot{n}_t = \left(n_t / \sqrt{T_{t,in}} \right) / \left(n_t / \sqrt{T_{t,in}} \right)_0$$

163 Fig. 4 illustrates the characteristic maps described by Eqs. (11) and (12).



164
 165 Figure 4: Characteristic maps for the turbines. a) Expansion ratio vs. reduced flow rate; b) Isoentropic
 166 efficiency vs. reduced flow rate.

167 Heat exchanger

168 The inter-refrigeration heat exchanger between LPC and HPC was modelled using energy balance equation
 169 and ε -NTU method [41]; considering a counter flow configuration the effectiveness was calculated as:

170
$$\varepsilon = \frac{1 - \exp[-NTU(1 - \chi)]}{1 - \chi \exp[-NTU(1 - \chi)]} \quad (14)$$

171 where

172
$$NTU = \frac{UA}{C_{\min}} \quad \chi = \frac{C_{\min}}{C_{\max}} \quad (15)$$

173 During the charging process effectiveness (14) was evaluated at each instant of time and the actual heat transfer
174 rate was computed as:

175
$$\Phi_{HEX} = \varepsilon \cdot C_{\min} (T_{in,h} - T_{in,c}) \quad (16)$$

176 From heat transfer rate Φ_{HEX} the air outlet temperature (i.e. the HPC inlet temperature) was obtained through
177 the energy balance equation for the heat exchanger.

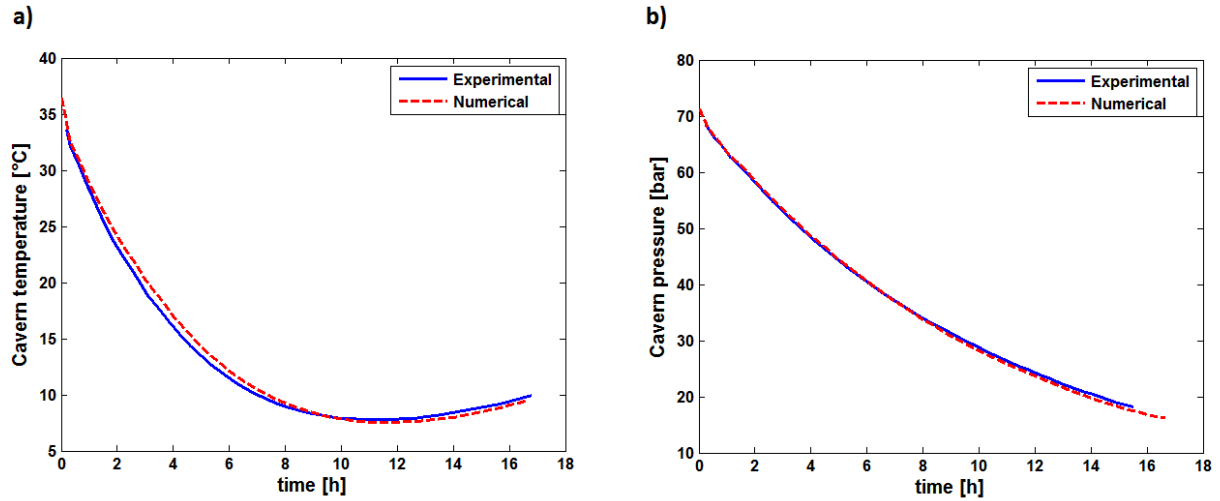
178 *Compressed air reservoir*

179 We employed a dynamic model to simulate the transient behaviour of temperature of air within the cavern.
180 The model consists of two ordinary differential equations that stem from energy balance and mass balance
181 equations for the air in the cavern [42]:

182
$$\frac{dT_r}{dt} = \frac{1}{m_r} \left[\left(1 - \frac{1}{k} \right) (\dot{m}_{in} T_{in} - \dot{m}_{out} T_r) + \frac{h_w A_w (T_w - T_r)}{c_{p,a}} \right] \quad (17)$$

183
$$\frac{dm_r}{dt} = \dot{m}_{in} - \dot{m}_{out} \quad (18)$$

184 In equation (17) the first term on the right hand side accounts for energy transfer due to injection/withdraw of
185 air from the cavern under the assumption that air leaves the cavern at the cavern's air temperature. The second
186 term quantifies the heat transfer between air and cavern's walls. Heat transfer coefficient h_w was evaluated as
187 indicated in [43]. Finally, pressure p within the compressed air reservoir was computed with ideal gas law
188 $p/\rho = \bar{R}T$. The model of the cavern was validated against the data gathered by Crotagino et al [8] from the
189 operation of the Huntorf plant. Figure 5 compares the experimental data and the numerical predictions for
190 cavern's temperature and pressure. The experimental data were recorder during a trial cavern discharge of 16
191 hours. As detailed in [8] the withdrawal rate was 417 kg/s for about 4 hours and then gradually decreased to
192 150 kg/s at the end of the test. The numerical results match the experimental data for the whole process and
193 correctly predicts the initial decrease of temperature – due to high withdrawal rate – and the final temperature
194 increase caused by heat transfer with the cavern's walls.



195

196 Figure 5: Validation of cavern model; numerical predictions against experimental data from Crotagino F. *et*
 197 *al.* [8].

198 *Packed bed thermal energy storage*

199 We adopted a non-equilibrium model to study heat transfer within the packed bed thermal energy storage
 200 (TES). Such an approach has been successfully employed in the literature by various authors [22,23,28,34]
 201 and it consists in a set of two energy balance equations, the first one for the air (subscript *a*) in the TES while
 202 the second one for the solid filler material (subscript *s*):

203
$$\varepsilon\rho_a c_{p,a} \frac{\partial T_a}{\partial t} + \varepsilon\rho_a c_{p,a} u_a \frac{\partial T_a}{\partial x} = k_a \frac{\partial^2 T_a}{\partial x^2} - h_v(T_a - T_s) - U_w(T_a - T_0) \quad (19)$$

204
$$(1 - \varepsilon)\rho_s c_{p,s} \frac{\partial T_s}{\partial t} = k_s \frac{\partial^2 T_s}{\partial x^2} + h_v(T_s - T_a) \quad (20)$$

205 In Eqs. 19 and 20 we assumed – as commonly done in the literature [22,23,28,34] – one dimensional heat
 206 transfer along the packed bed length *x*. Void fraction ε of the bed was evaluated as function of the ratio particle
 207 diameter d_p to packed bed diameter *D* [34]:

208
$$\varepsilon = 0.375 + 0.17 \frac{d_p}{D} + 0.39 \left(\frac{d_p}{D} \right)^2 \quad (21)$$

209 Air velocity u_a was evaluated at each instant of time starting from compressor mass flow rate, during charge,
 210 and turbine mass flow rate during discharge. Uniform velocity throughout the TES transversal cross section
 211 was considered. The thermal conductivity k_s of the bed was evaluated by means of the Zehner-Bauer-
 212 Schlunder model [44].

213 The second term on the right hand side of Eq. (20) accounts for the heat transfer between the air and the solid
 214 particles in the thermal storage system. The volumetric heat transfer coefficient h_v was computed using the
 215 Coutier's correlation [25,34]:

216
$$h_v = 700 \left(\frac{G}{d_p} \right)^{0.76} \quad (22)$$

217 where G is the mass flux ($\text{kg s}^{-1} \text{m}^{-2}$) flowing through the packed bed thermal storage system. The last term on
 218 the right hand side of Eq. (20) quantifies the heat loss toward the ambient at temperature T_0 . We attributed the
 219 heat loss entirely to the fluid phase (Eq. 20) since separate correlations are not available in the literature and
 220 experiment cannot distinguish properly between phases [34]. Heat transfer coefficient U_w was determined
 221 considering heat transfer through a multi-layer cylindrical wall [41]. Table 4 summarizes the major parameters
 222 of the TES system. The diameter D and height H of the TES system were obtained through a preliminary
 223 design on the basis of data in Tables 1 and 2 together with charge/discharge cycle of Fig. 2. Such data allows
 224 to estimate heat to be stored and thus the geometrical dimensions of the TES system. Such dimensions are in
 225 line with those reported in [19, 45], although other arrangements, such as multiple TES in parallel/series could
 226 be also considered.

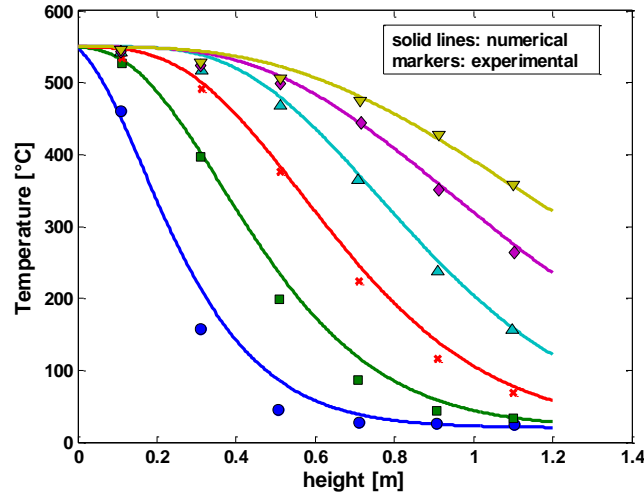
227 Table 4. Input parameters for TES model

Property	Formulation
ρ_s (kg/m^3)	2911 [22]
$c_{p,s}$ (J/kg K)	$A(B + CT + B/T^2)$ [47]
k_s (W/m K)	Zehner-Bauer-Schlunder model [44]
d_p (m)	0.02 [34]
H (m)	22
D (m)	20

228

229 To validate the model the numerical predictions were compared with experimental results obtained by Meier
 230 et al. [26]. The researchers studied a lab scale packed bed thermal energy storage and recorded temperature
 231 along the packed bed during charge. The major parameters of the experimental set up considered by Meier et
 232 al. are available in [26,34]. Our packed bed model, comprising Eqs. 19-22, was run in standalone mode using
 233 mass flow rate and thermos-physical properties available in [26,34] as input parameters. Clearly, in our
 234 validation study we adopted the same packed bed diameter D and length L considered by Meier et al [26].

235 Figure 6 compares the temperature profile along the TES predicted by our model and the experimental data
 236 from [26] at different instants of time. The comparison demonstrates that the model is capable of predicting
 237 both temperature and position of the thermal front with good accuracy. Discrepancy between experiments and
 238 simulation can be attributed to the small ratio $D/d_p = 7.5$ considered in [26]: when particle diameter d_p is
 239 relatively large compared to packed bed diameter D a non-negligible fraction of the air mass flow rate passes
 240 near the walls of the packed bed, thus it does not contribute to heat transfer with the filling material. Therefore,
 241 the accuracy of the model, which is already satisfactory for a small lab scale device, will further improve when
 242 a full scale system is considered.



243

244 Figure 6: Validation of packed bed model; Numerical predictions against experimental data from Meier et al.
 245 [26].

246 *Performance indicators*

247 The round trip efficiency and the thermal storage efficiency were used to assess the performance of the whole
 248 A-CAES plant and the thermal energy storage system. The round trip efficiency for each charge/discharge
 249 cycle was calculated as:

$$250 \quad \eta_{cycle} = \frac{E_{out}}{E_{in}} = \frac{\int_0^{\Delta t_d} W_t dt}{\int_0^{\Delta t_c} W_c dt} \quad (23)$$

251 The time integration is performed over each charging period (Δt_c) and discharging period (Δt_d).

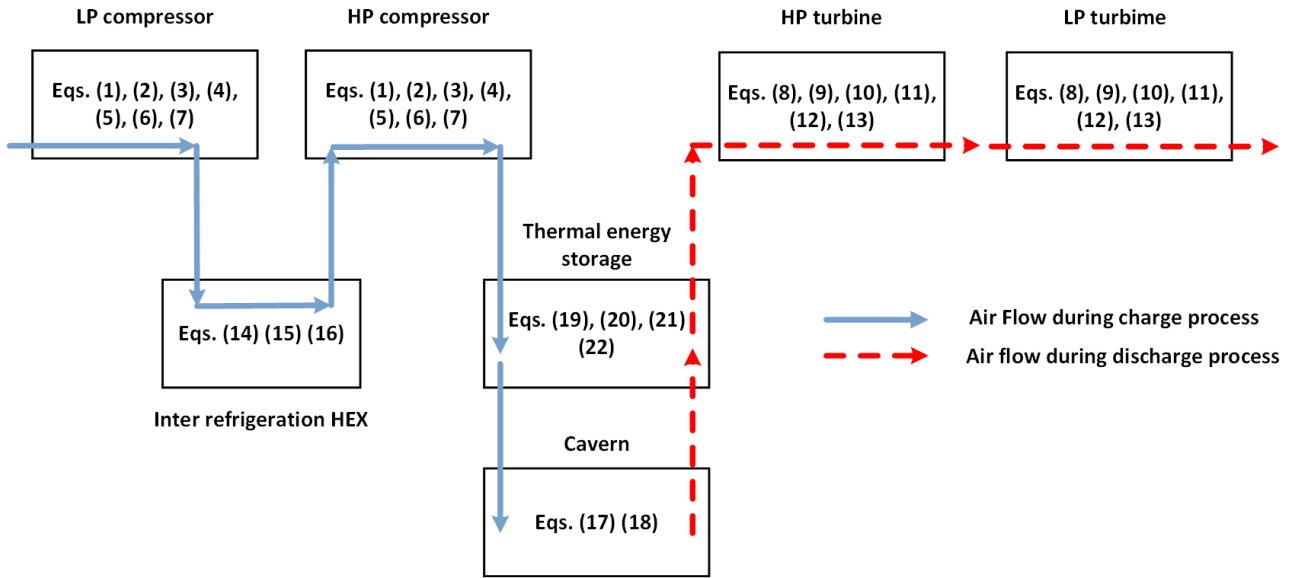
252 The performance of the TES system was assessed through the thermal storage efficiency defined as follows:

$$253 \quad \eta_{th} = \frac{\int_0^{\Delta t_d} \dot{m}_t (h_4 - h_0) dt}{\int_0^{\Delta t_c} \dot{m}_c (h_1 - h_0) dt} \quad (24)$$

254 Where h_4 is the specific enthalpy of air at the outlet of the TES system during discharging while h_1 is the
 255 specific enthalpy of air the inlet of TES system during charging.

256 *A-CAES plant simulations*

257 The previous equations were implemented in Matlab/Simulink to simulate the entire A-CAES plant. The block
 258 diagram of Fig. 7 shows how the sub-systems interact during calculation for charging and discharging
 259 processes. The equations were solved using 4th order Runge Kutta method with variable time step. Two distinct
 260 sub-sets of equations were solved depending if the plant operates in charge or discharge mode. The interactions
 261 between the components of the plant lead to two sub-sets of coupled equations, as clarified by the process flow
 262 indicated by the arrows in Fig. 7. The arrows in the figure indicates which components (blocks), and therefore
 263 the corresponding equations, involved during simulation of charge and discharge processes. Separate blocks
 264 exchanging information (mass flow rate, pressure and temperature) at each instant of time were implemented
 265 in Simulink.



266

267

Figure 7: Block diagram for the whole A-CAES plant model.

268

4 Results and Discussion

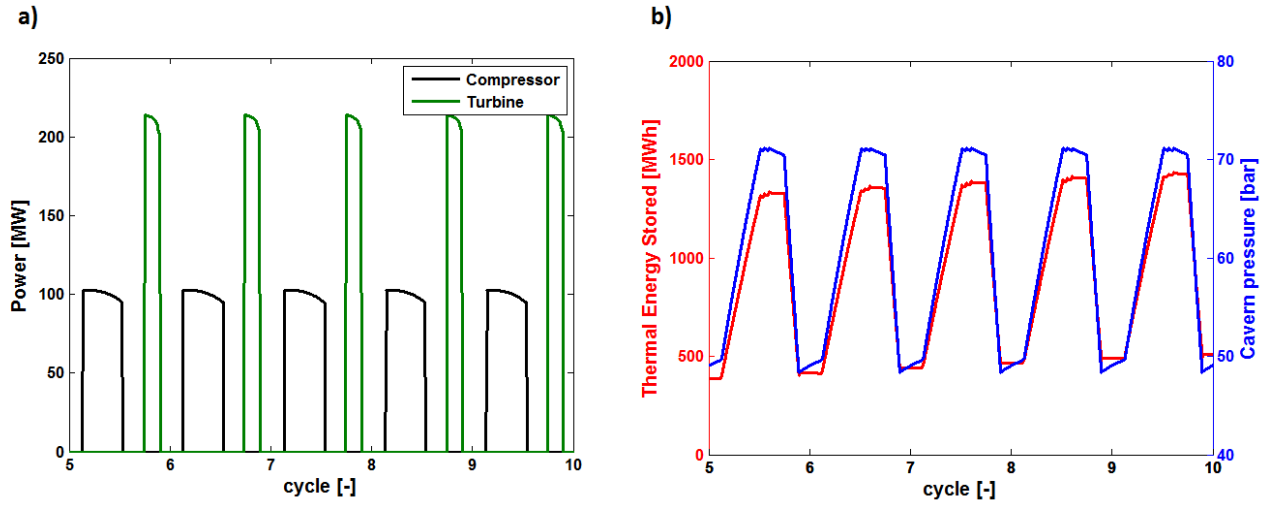
269

The simulations provide results for each component depicted in Fig. 1 for 30 consecutive charge/discharge cycles. A subset of these results are shown in Fig. 8 to clarify the plant operation; detailed results for each component of the plant are presented in the following subsections. Fig. 8a plots the power input/output for the A-CAES plant. Both compression train and expansion train operate around the corresponding rated power (100 MW/220MW); the variations in compression power and expansion power during charge stage and discharge stage are due to off-design operating conditions which will be detailed in sections 4.2 and 4.3. Fig. 8b shows the state of charge for the thermal store and the compressed air reservoir. The stored thermal energy and the air pressure show a similar time pattern, since both follow charge/discharge cycles; 940 MWh_{th} are stored in the TES on average (see Table 5) while TES efficiency, as defined by Eq. (24), is 93%. Cavern pressure spans the range 48-71 bar. Pressure variation occurs also during idle stage: after each discharge process cavern pressure increases from ~ 48 bar to about 50 bar due to heat transfer from cavern walls to compressed air, the latter being relatively cold because of the withdraw process during discharge. A similar effect was also pointed out in [46]. During idle after each charge, the pressure in the cavern slightly drops before the next discharge. In this case heat flows from the compressed air to the cavern wall cooling the mass of air in the cavern and thus leading to a reduction of pressure.

284

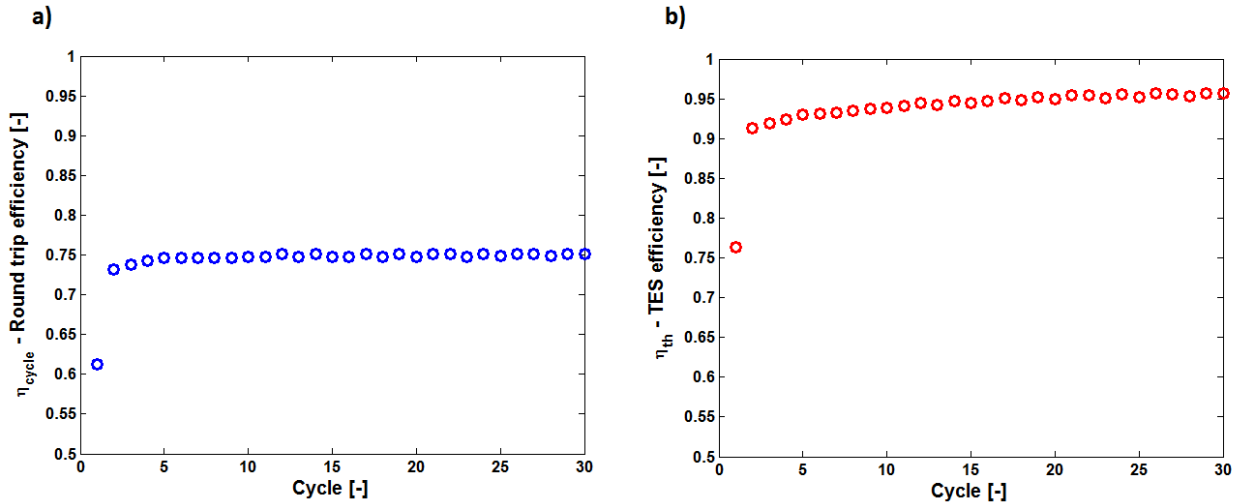
Figure 9 displays the round trip efficiency and thermal storage efficiency over 30 cycles. Both efficiencies reach a stable value after an initial increase during the first operating cycles. The key quantity here is the efficiency of the thermal storage system: as soon as TES starts to operate in an effective way the overall performance significantly improves, which shows how relevant is to integrate carefully the thermal storage with the remaining part of the system. Maximum TES efficiency occurs when static cycling operating conditions are established in the thermal storage as explained in the next section. Clearly, the predicted value for of roundtrip efficiency are affected by the value of parameters used in the model. We adopted, whenever possible, values commonly used in the CAES literature and that led to accurate results when compared with experimental data. We estimated, by varying turbine isentropic efficiency between 0.8 and 0.88, that changes in roundtrip efficiency stays within $\pm 10\%$.

293



294

295 Figure 8: A-CAES plant performance between 5th and 10th operation cycle. a) Compression train power during
 296 reservoir charge and turbine power output during discharge. b) Thermal energy stored in the sensible TES (left
 297 axis) and reservoir pressure variation (right axis) due to injection/withdraw of compressed air.



298

299 Figure 9: Efficiency of A-CAES plant. a) Round trip efficiency according to Eq. (23); b) Efficiency of the
 300 thermal energy storage system (Eq. 24).

301

Table 5. A-CAES performance for full load charging/discharging

Quantity	Value
Number of cycles (-)	30
Round trip efficiency η_{cycle} (-)	74%*
Total output energy (MWh _e)	22100
Charge time (h)	9.1*
Discharge time (h)	3.3*
Thermal energy stored (MWh _{th})	940*
Thermal energy storage efficiency η_{th} (-)	93%*

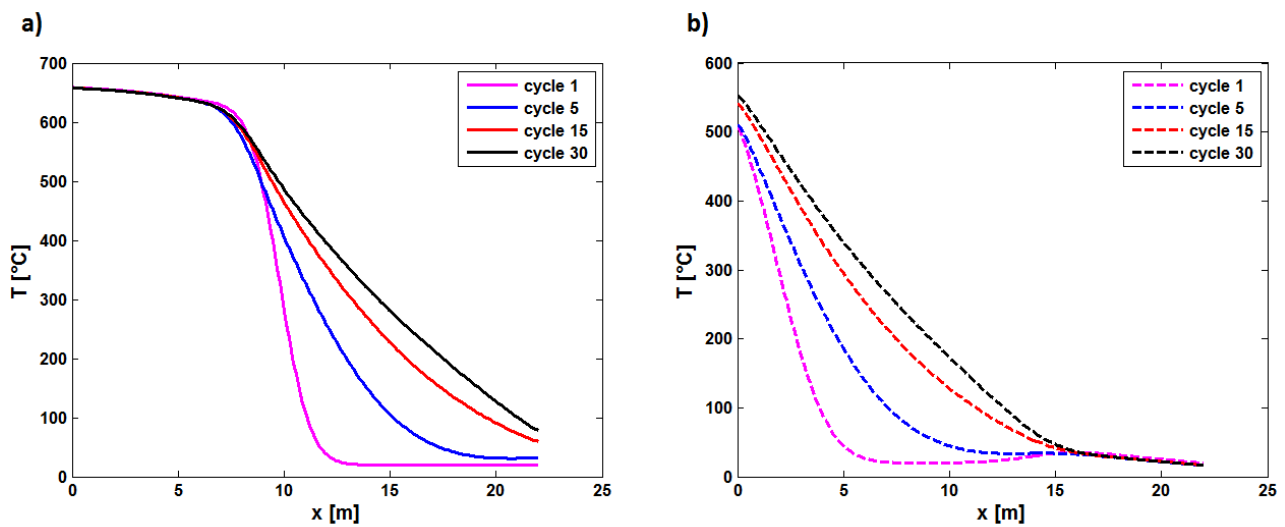
* Averaged value over 30 cycles

302

303 4.1 Thermal energy storage (TES) system

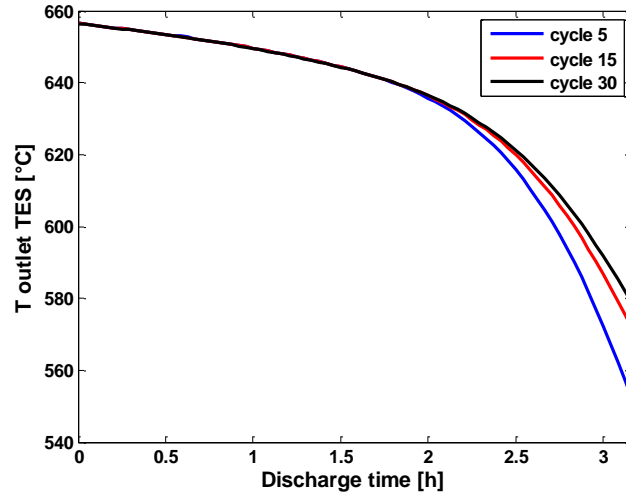
304 Figure 10 presents the temperature profile within the TES system and shows how the temperature profile varies
 305 from cycle to cycle. Figure 10a shows temperature after charge ($t = 16\text{h}$ within each cycle). Two key features

306 should be noticed: the position of the thermal front and how the temperature evolves, after a sufficient number
 307 of cycles, toward a cycling stationary profile. After cycle 1 the temperature shows a thermal front around $x =$
 308 11 m that extends for about 10% of the TES length. The ideal operation of the TES system, as illustrated in
 309 [28,31], would preserve the thermal front as sharp as possible from cycle to cycle, while each charge/discharge
 310 would consist in such sharp front travelling back and forth from $x = 0$ to $x = H$. Thermal degradation of the
 311 front [28] prevents a practical implementation of the ideal TES operation, in fact after 5 cycles thermal front
 312 broadens up to 50% of TES length. Therefore, the thermal store actually operates very similarly to a
 313 regenerator: air is gradually cooled during charge, while it is gradually heated – from TES inlet to TES outlet
 314 – during discharge. Such an operation mode leads to stationary cycling operating conditions, where two
 315 stationary temperature profiles occur after charge and discharge (see cycle 30 in Fig. 10). Stationary profile
 316 slightly decreases from $x = 0$ m to $x = 10$ m due to increase in air outlet temperature from HP compressor
 317 during charge. During discharge, air withdrawn from the cavern is slightly above ambient temperature; this
 318 causes the hump at $x = 15$ m illustrated in Fig. 10b.



319
 320 Figure 10: Temperature profile along the length of the thermal energy storage (TES) system. a) Temperature
 321 profiles after charging; b) Temperature profiles after discharge. Cycling operating conditions establish after
 322 20 cycles of charging/discharging.

323 The time evolution of air at the outlet of TES system – corresponding T_4 in Fig. 1 – is presented in Fig 11. The
 324 outlet temperature stays within a range of 20°C for about 67% of the discharge time; such an operating
 325 condition corresponds to the time necessary for the flat portion of the TES temperature profile ($x < 10$ m in
 326 Fig. 10a) to leave the thermal store during discharge. During the last stage of discharge the outlet temperature
 327 drops of about 15%, as the degraded thermal front exits the thermal store. A more marked drop occurs during
 328 the first cycles because stationary temperature profile is not established yet in the TES system. Air outlet
 329 temperature from the packed bed storage coincides with the HP turbine inlet temperature; thus, any variation
 330 of T_4 from design point detracts the performance and efficiency of the expansion train, as explained in Sect.
 331 4.3. These results presented here can help CAES operators to conceive optimal operating strategy to reduce
 332 such undesired off-design conditions.



333

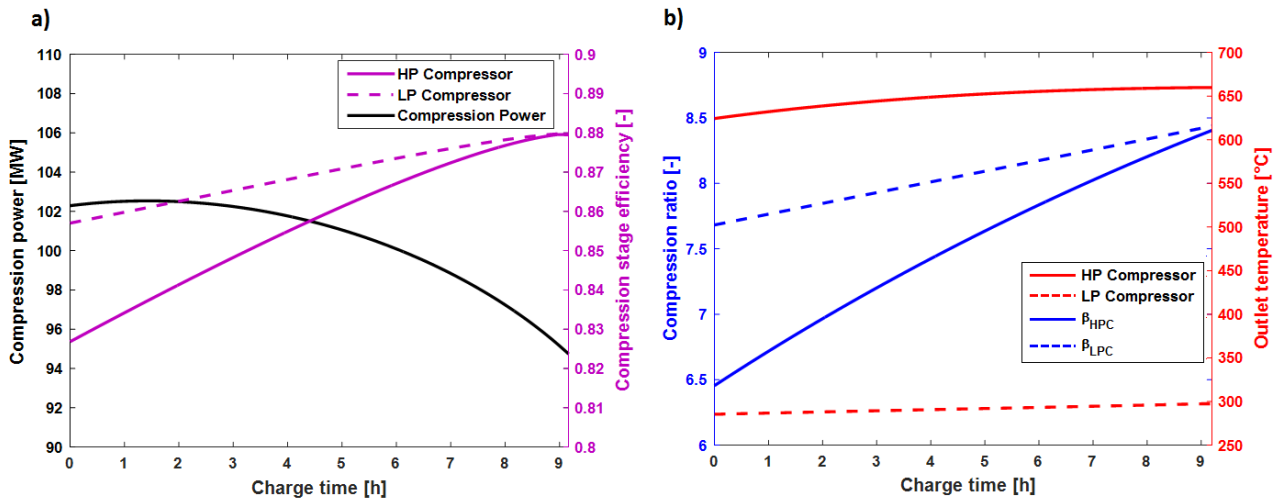
334 Figure 11: Temperature profile along the length of the thermal energy storage (TES) system. a) profiles after
 335 charging; b) profiles after discharge. Cycling operating conditions establish after 20 cycles of
 336 charging/discharging.

337 4.2 Compression train

338 The performance of low pressure compressor (LPC) and high pressure compressor (HPC) significantly depart
 339 from nominal condition, as both compressors operate off-design during charging. In fact, the pressure of air in
 340 the cavern constantly increases during charge, causing an increase also in the total pressure ratio experienced
 341 by the compressor train. As a consequence, compressors' operating point moves along characteristic curve
 342 (Fig. 3) from low to high pressure ratio. Figure 12 lucidly summarizes how the compression train operates
 343 during charge. At the beginning of charging the LPC performs the majority of the compression work as LP
 344 compression ratio is $\sim 16\%$ larger than the HP one. As pressure in the cavern rises, both β_{LPC} and β_{HPC}
 345 increment up to the corresponding design values, which is achieved only at the end of charge. As charging
 346 starts $\beta_{LPC} = 7.6$ and $\beta_{HPC} = 6.5$, thus LPC compression ratio and HPC compression ratio are respectively 10%
 347 and 22% lower than the design value. As a result, the minimum isentropic efficiency of compressors occurs
 348 at the begin of each charge as shown in Fig. 12a. Figure 12b shows that at the end of charging the HP
 349 compressor outlet temperature is 660°C while – from Fig 11 – we found at the beginning of discharge air
 350 leaves the TES systems at 656°C . Such a difference between the two temperatures is due to finite heat transfer
 351 between the air stream and the rocks within the TES. However, the difference is very limited due to the good
 352 thermal contact (large heat transfer area) between air and TES filling material.

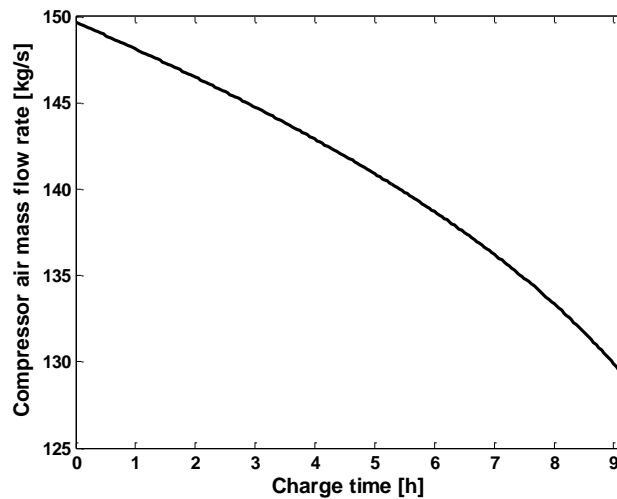
353 The compression power shows a maximum around $t = 2\text{h}$ which can be explained from the behaviour of mass
 354 flow rate (Fig. 13) and compression ratios (Fig. 12b). The combination of decreasing trend for \dot{m}_c with an
 355 increasing trend for β_{LPC} , β_{HPC} brings a maximum in compression power W_c (Eq. 3). The mass flow rate
 356 monotonically decreases during charge because the operating point of compressors (Fig. 3) shifts from low
 357 compression ratios, so high mass flow rate \dot{G}_c , to high compression ratio and lower mass flow rate. On the
 358 other hand, the compression ratio monotonically increases during charge as cavern pressure rises.

359



360

361 Figure 12: Compression train performance during charge. a) Compression power and isoentropic efficiency of
 362 high pressure and low pressure compressors. b) Compression ratio and air outlet temperature for high and low
 363 pressure compressors. Compressor train operates under off-design conditions except at the end of the charging
 364 process.



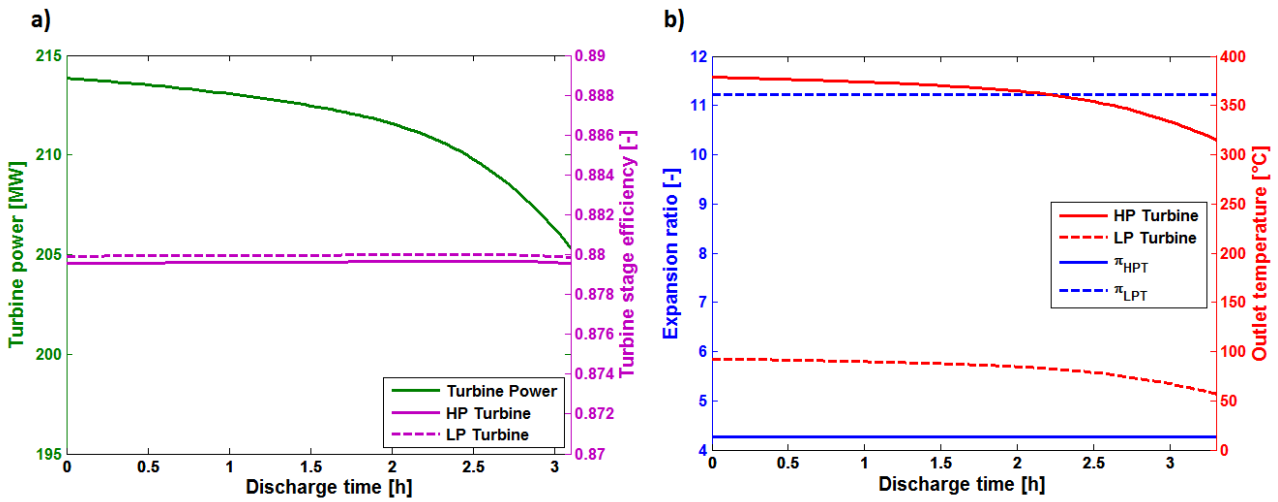
365

366 Figure 13: Compressor air mass flow rate during charge. During charge compression ratio increases (Fig.
 367 12b) consequently mass flow rate diminishes as compressor operation point moves along the characteristic
 368 curve (Fig. 3).

369 4.3 Expansion train

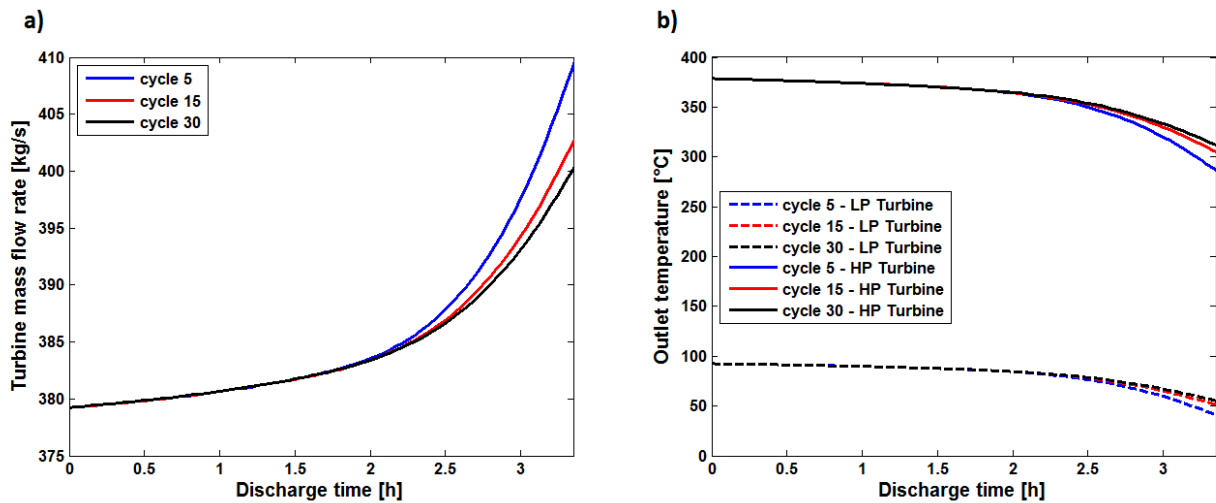
370 The expansion train operates under constant expansion ratio – due to the throttling valve (Fig. 1) – but with
 371 variable inlet temperature of air coming from the thermal energy storage system (Fig. 11). As a results
 372 departure from design condition are limited in comparison with the compression train, as illustrated in Fig. 14.
 373 Both high pressure turbine (HPT) and low pressure turbine (LPT) perform at design isoentropic efficiency for
 374 the entire discharge process. The power output drops of about 5% during discharge due to a combined effects
 375 of decrease in inlet temperature (Fig. 11) and variation of the turbine mass flow rate (Fig. 15). Although the
 376 turbine mass flow rate increases, as depicted in Fig. 15a, the drop in inlet temperature (Fig 11) dominates the
 377 behaviour of turbine power, resulting in a reduction of power output from the A-CAES plant. This shows how
 378 important is to conceive and operate the thermal storage system in an optimal way, since TES performance
 379 reverberate onto the global performance of the plant. Variations of the turbine mass flow rate are also caused

380 by reduction of air inlet temperature: according to the Flügel formula (Eq. 11) at constant expansion ratio we
 381 have $\dot{m}_t \propto 1/\sqrt{T_{inlet}}$. Finally, the decrease of the air outlet temperature from HPT and LPT shown in Fig. 15b
 382 stems directly from the reduction in the inlet temperature.



383

384 Figure 14: Expansion train performance during discharge. a) Turbine power and isoentropic efficiency of high
 385 pressure and low pressure turbine. b) Expansion ratio and air outlet temperature for high and low pressure
 386 turbine. Expansion train operates near design conditions for most of discharge process because of constant
 387 inlet pressure.



388

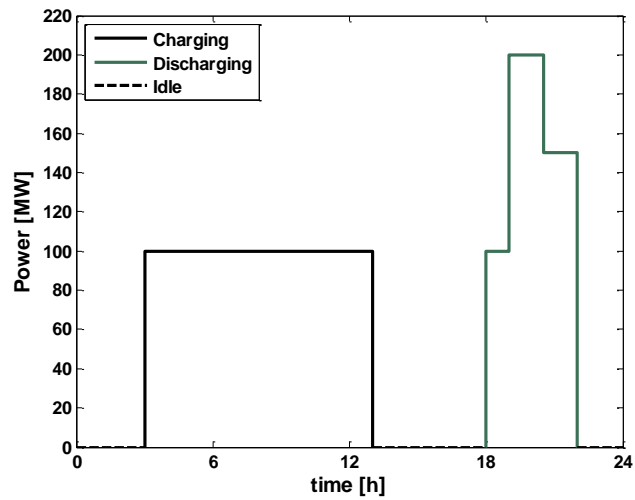
389 Figure 15: Variation of turbine operation over charge/discharge cycles. a) Turbine mass flow rate b) Air outlet
 390 temperature from low pressure and high pressure turbine. Variation of turbine inlet temperature (Fig. 11) leads
 391 to increase of flow rate due off-design conditions.

392 4.4 Partial load operation

393 The operation of CAES systems for peak shaving, minute reserve, or compensation of fluctuation in wind
 394 power likely involves partial load operation during discharging [37]. The model we developed allows us to
 395 study A-CAES performance for partial load operating cycle. We considered the cycle of Fig. 16 to show how
 396 partial load conditions may detriment A-CAES performance. In the view of peak shaving operation we
 397 considered a discharge cycle that last four hours (as in case of Fig. 2) but at three different loads. This mimics
 398 operating condition that may realistically occurs, as presented in [8,43]. The power output is controlled by

399 adjusting the inlet pressure for HP turbine by throttling air flow from the cavern. Table 6 summarizes the
 400 results for this operation mode.

401 Figure 17 shows the performance indicators for A-CAES plant and TES system. Round trip efficiency
 402 detriments due to smaller power output while TES is marginally affected by partial-load operation which
 403 causes variation of air flow through the TES as detailed below. As the inlet pressure varies with the load, HP
 404 and LP expansion ratios adjust accordingly (Fig 18). Maximum relative variation of π_{HPT} is nearly 40% which
 405 causes non-negligible changes in the corresponding isentropic efficiency. The outlet temperature from the
 406 turbine stages (Fig. 18a) varies following the changes in the expansion ratios. The outlet temperature drops
 407 toward the end of discharging cycle since the temperature of air from TES reduces, as previously illustrated
 408 for Fig. 11. Cycle-to-cycle variations can be seen in Fig. 19, as stationary temperature profile establishes within
 409 the thermal energy storage system.

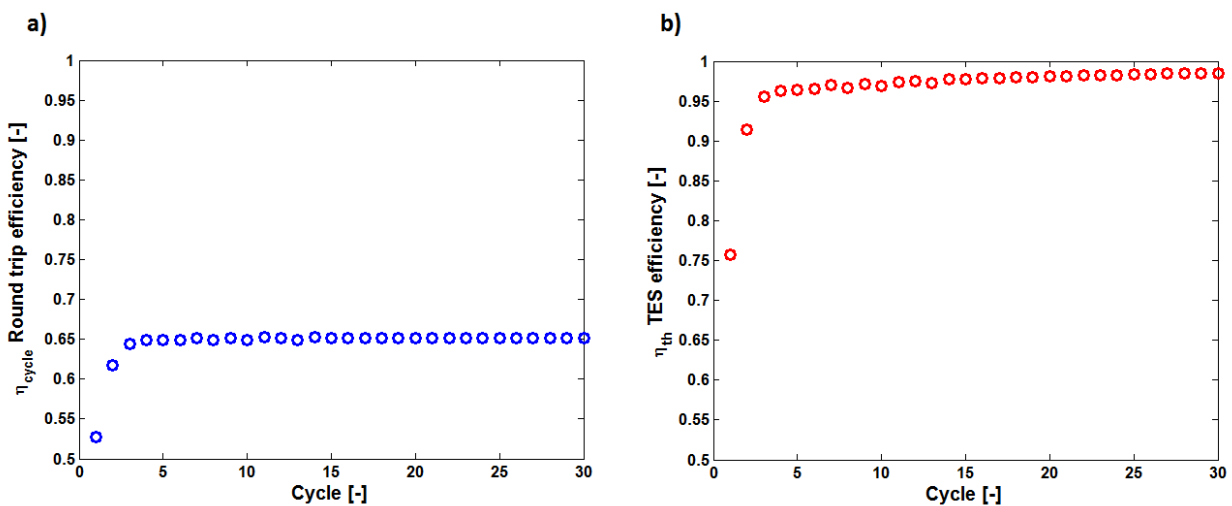


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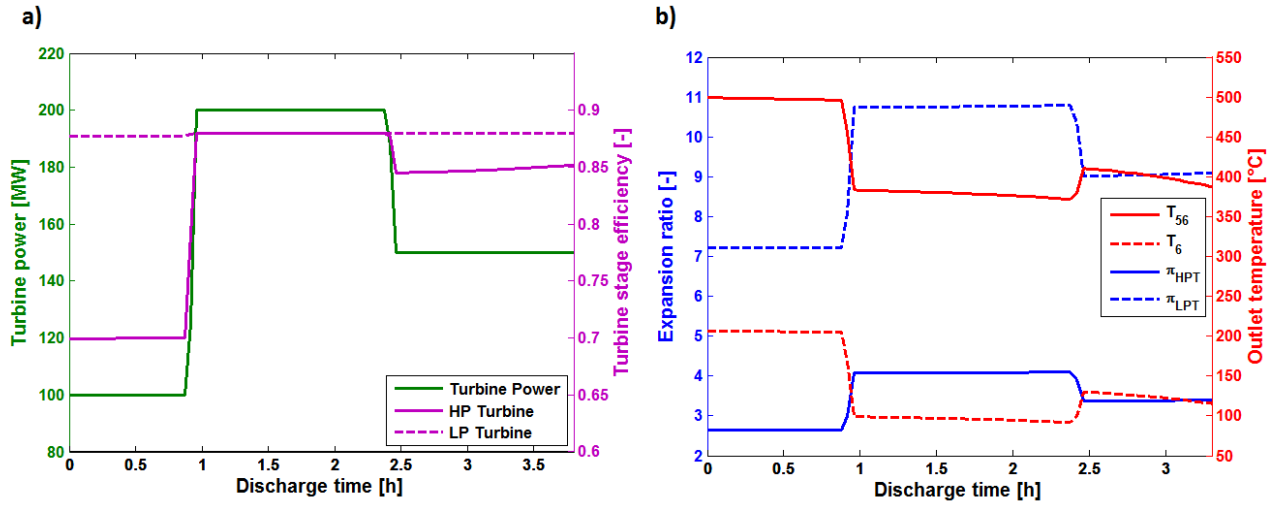
Figure 16: Charge and discharge cycle – partial load case.

412



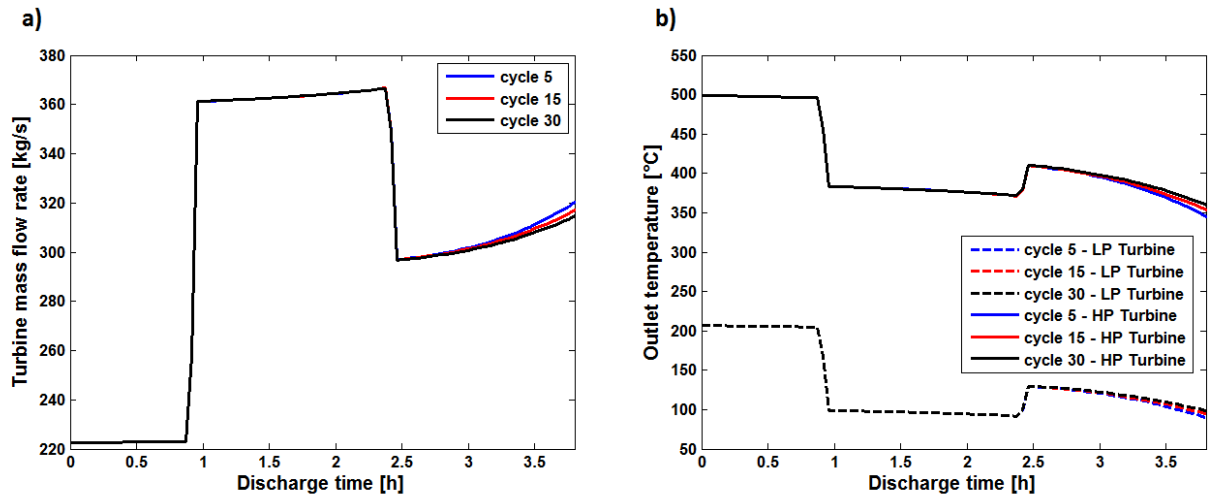
413

414 Figure 17: Efficiency of A-CAES plant under partial load operation. a) Round trip efficiency according to Eq.
 415 (23); b) Efficiency of the thermal energy storage system (Eq. 24).



416

417 Figure 18: Expansion train performance during partial load discharge. a) Turbine power and isoentropic
 418 efficiency of high pressure and low pressure turbine. b) Expansion ratio and air outlet temperature for high and
 419 low pressure turbine.



420

421 Figure 19: Variation of turbine operation over charge/discharge cycles at partial load. a) Turbine mass flow rate
 422 b) Air outlet temperature from low pressure and high pressure turbine.

423

Table 6. A-CAES performance for partial load operation

Quantity	Value
Number of cycles (-)	30
Round trip efficiency η_{cycle} (-)	64%*
Total output energy (MWh _e)	18900
Charge time (h)	8.5*
Discharge time (h)	3.8*
Thermal energy stored (MWh _{th})	860*
Thermal energy storage efficiency η_{th} (-)	96%*

* Averaged value over 30 cycles

424

425 5 Conclusions

426 In this paper we developed for the first time a fully dynamic and off-design performance model of an A-CAES
 427 plant with a packed bed thermal energy storage (TES) system. This was possible by integrating together

428 algebraic and differential sub-models that detail the transient features of the thermal storage, the cavern, and
429 the compression/expansion stages, which is a novelty proposed in this work.

430 Both design and off-design charging/discharging cycles were studied for the specific A-CAES plant
431 considered. The results indicate that under nominal charging/discharging a round trip efficiency exceeding
432 70% can be achieved when TES efficiency rises above 90%. The link between device performance with plant
433 performance was elucidated. In fact we can conclude that: i) maximum round trip efficiency occurs when
434 cycling stationary temperature profiles establishes in the packed bed TES; ii) A-CAES performance detriments
435 toward the end of each discharging cycle due to degradation of the thermal front within the thermal store; iii)
436 reduction of air outlet temperature from TES system causes the turbine to operate in off-design conditions
437 leading to an increase of flow rate; iv) the compressors operate under strong off-design conditions which also
438 affect temperature profile in thermal storage system.

439 In summary, we showed that the linking device dynamic performance with system performance is a necessity,
440 since modern energy storage systems present a strong tendency toward transient operation. We achieved such
441 a goal, for the first time for A-CAES, with the work presented in this paper.

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445 (EP/L014211/1)

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